

Impact of new background models on GNSS orbit determination

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6th International Colloquium - Scientific and Fundamental Aspects
of GNSS / Galileo, 25-27 October 2017,
Valencia, Spain

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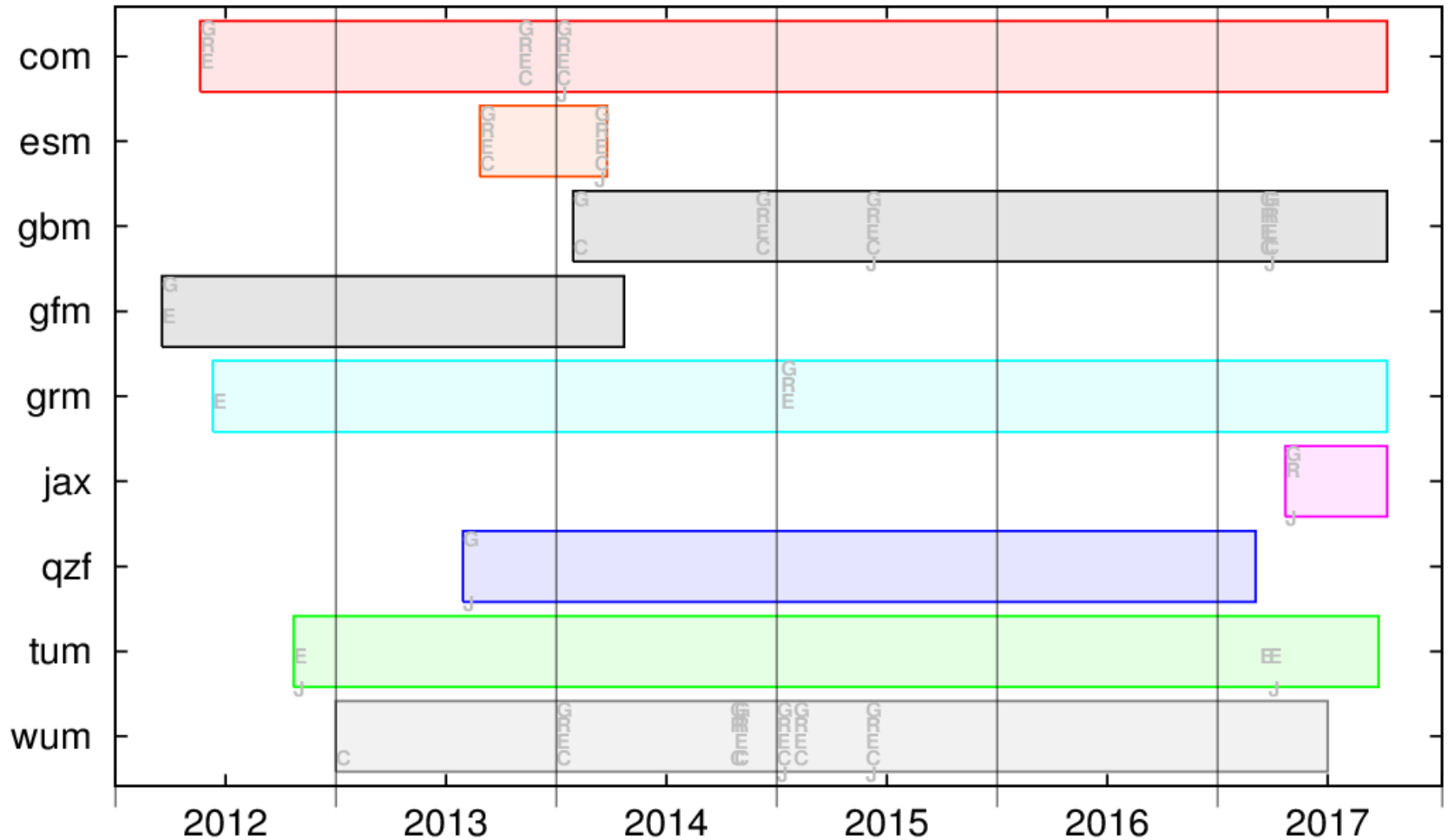
CODE MGEX orbit solution

GNSS considered:	GPS + GLONASS + Galileo + BeiDou (MEO+IGSO) + QZSS (>80 SV)
Processing mode:	Post-processing (\approx 2 weeks latency)
Timespan covered:	GPS-weeks 1689 - today
Number of stations:	140 (GPS), 125 (GLONASS), 95 (Galileo); 70 (BeiDou); 40 (QZSS)
Processing scheme:	Double-difference network processing (observable: phase double differences)
Signal frequencies:	L1+ L2 (GPS + GLO+ QZSS); E1 (L1) + E5a (L5) Galileo; B1 (L2) + B2 (L7) BeiDou
Orbit characteristic:	3-day long arcs; SRP: ECOM / ECOM2 (since 2015)
Reference frame:	IGb08 (until week 1934); IGS14 (since week 1934)
IERS conventions:	IERS2003 (until 1705); IERS2010 (since week 1706)
Product list:	Daily orbits (SP3; 900 => 300s since 1962) and ERPs
Distribution:	ftp://cddis.gsfc.nasa.gov/gnss/products/mgex/ and ftp://ftp.aiub.unibe.ch/CODE_MGEX/ comwwwwd.???.Z => COD0MGXFIN_YYYYDDD...gz (since week 1962)
Designator:	week 1962)

CODE MGEX clock solution

GNSS considered:	GPS + GLONASS + Galileo + BeiDou + QZSS (>80 SV)
Processing mode:	Post-processing (\approx 2 weeks latency)
Timespan covered:	GPS-weeks 1710 - today
Number of stations:	140 (GPS), 125 (GLO), 95 (Galileo); 70 (BeiDou); 40 (QZSS)
Processing scheme:	Zero-difference processing (observable: code+phase undifferenced)
Signal frequencies:	L1+ L2 (GPS + GLO+ QZSS); E1 (L1) + E5a (L5) Galileo; B1 (L2) + B2 (L7) BeiDou
A priori information:	Orbits, ERPs, coordinates, and troposphere from CODE MGEX orbit solution introduced as known
Reference frame:	IGb08 (until week 1934); IGS14 (since week 1934)
IERS conventions:	IERS2010
Product list:	Epoch-wise (30s) satellite and station clock corrections in daily clock RINEX files; daily inter-system biases for mixed stations in Bernese DCB-, BIAS-SINEX-, OSB-format ftp://cddis.gsfc.nasa.gov/gnss/products/mgex/ and ftp://ftp.aiub.unibe.ch/aiub/CODE_MGEX/
Distribution:	

MGEX products availability

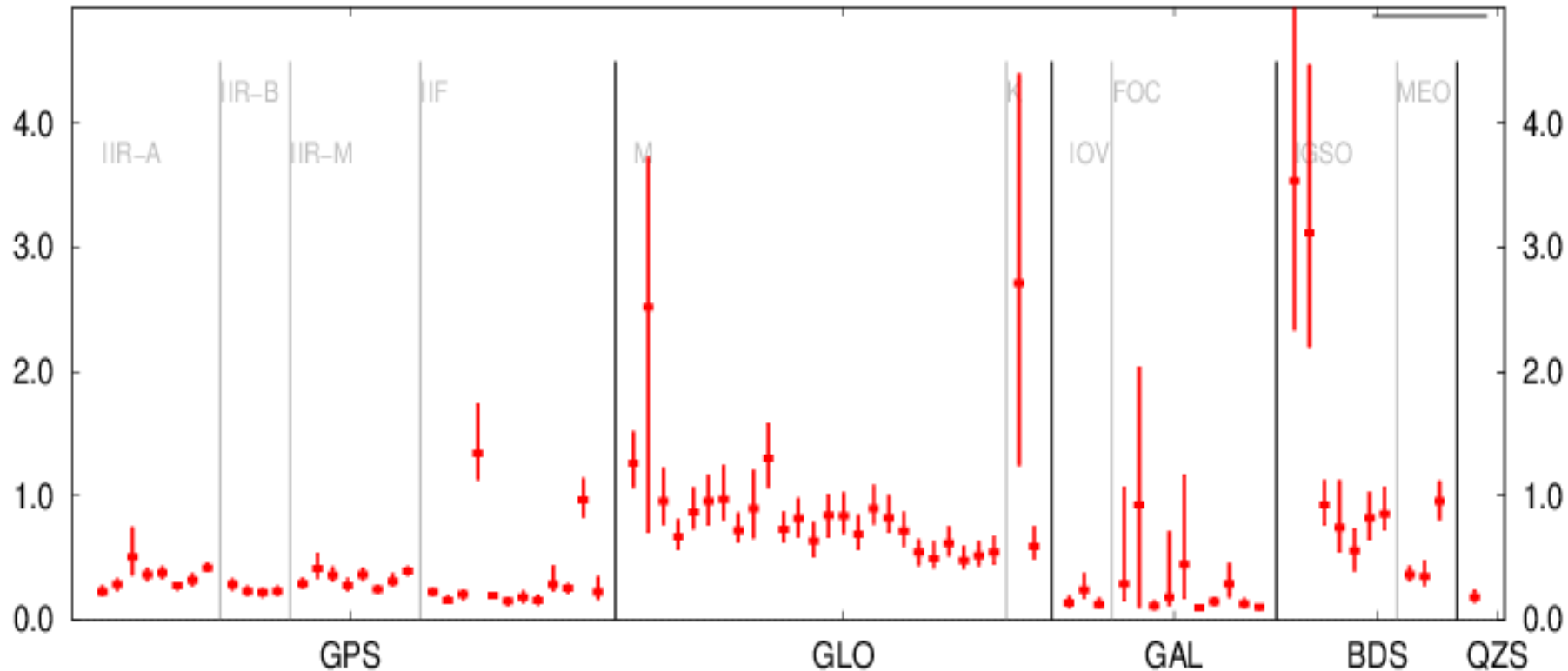


Status: October 2017

Satellite system IDs according to the content of the precise orbit files at <ftp://cddis.gsfc.nasa.gov/pub/gps/products/mgex/>

COM clock validation 2016: daily linear fit

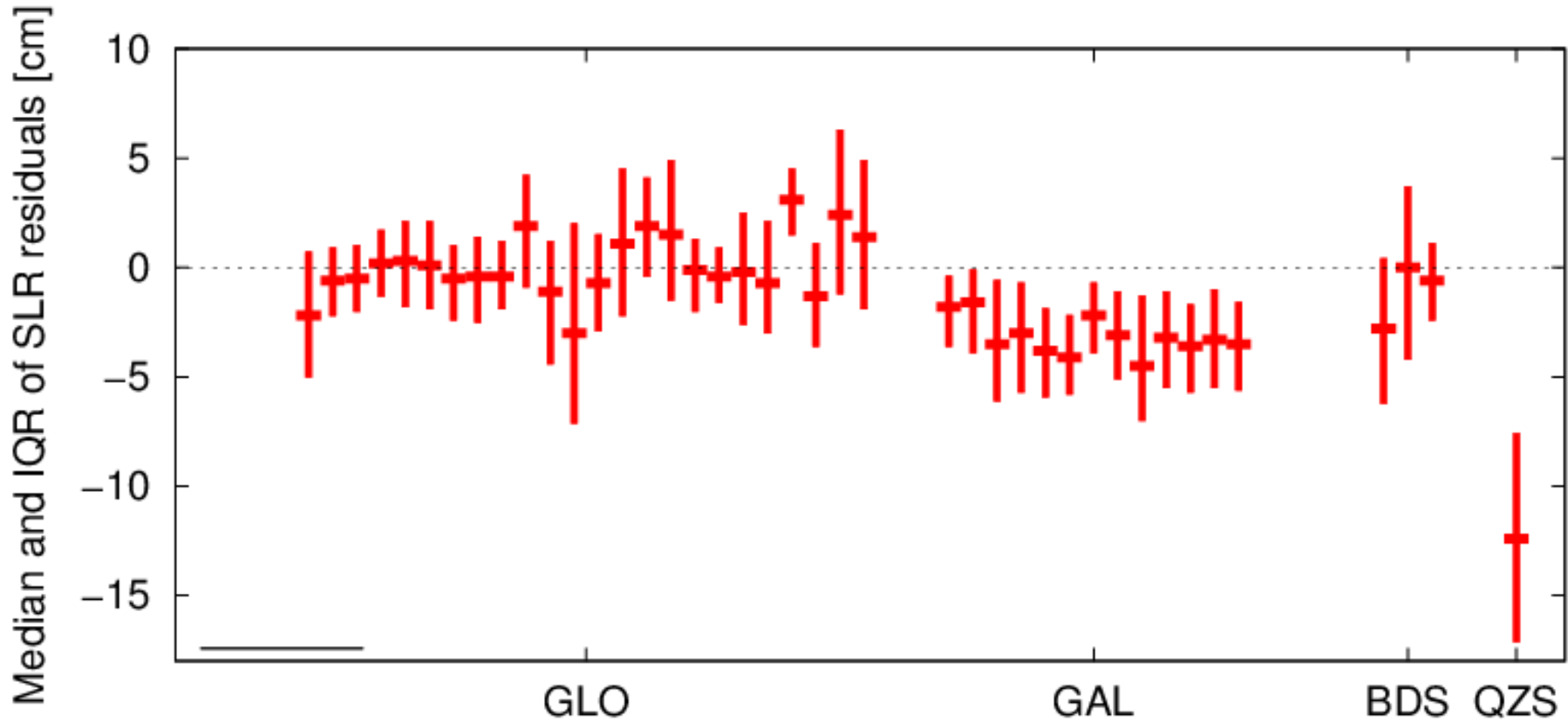
(Median and IQR; satellites in eclipse or normal mode are not considered)



- ⇒ Galileo PHM, QZS-1, most GPS IIR and IIF: excellent clocks (even suited for orbit validation)
- ⇒ Some GPS IIF, GLONASS, Galileo RAFS: worse (RMS: 0.5 ns or bigger)
- ⇒ BeiDou: mixed performance

COM orbit validation 2016: SLR residuals

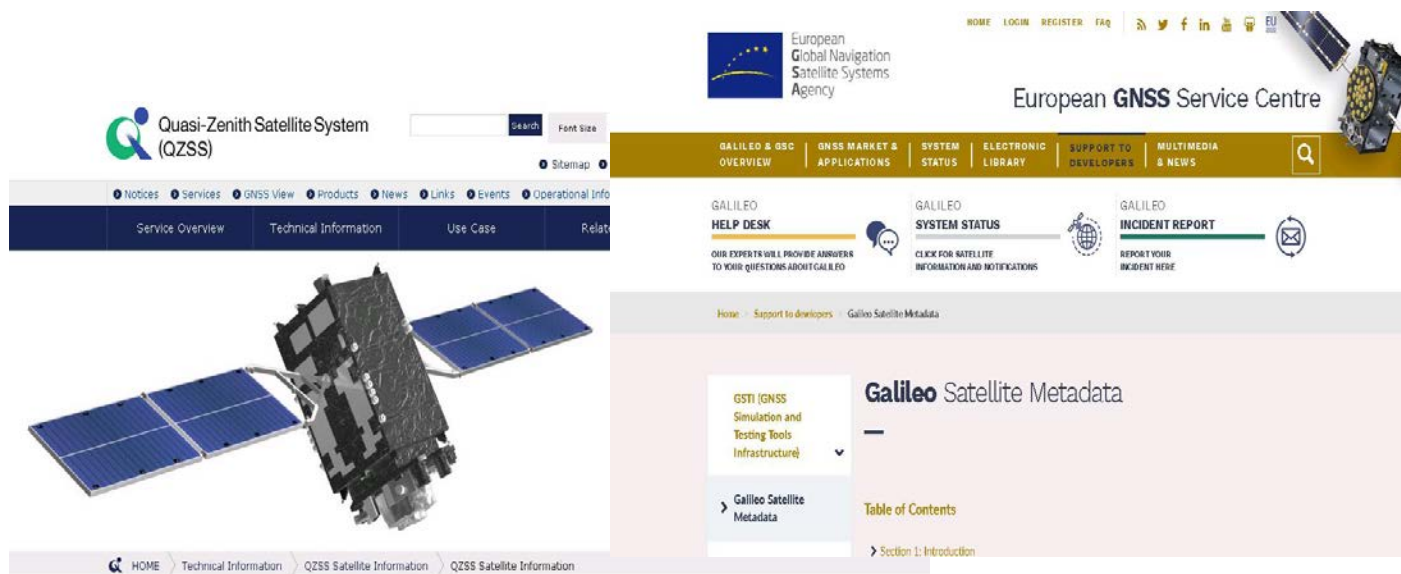
(Median and IQR; satellites in eclipse or normal mode are not considered)



- ⇒ Significant SLR offsets for Galileo and QZSS due to orbit model deficiencies with impact on radial orbit component (respectively scale)
- ⇒ Model improvements are needed (e.g., ANTEX, albedo, antenna thrust)
- ⇒ See doi 10.1007/s00190-016-0968-8 for more information

Release of satellite meta data ...

- Missing satellite meta data is a limiting factor for accuracy of estimated orbits and clocks, therefore ...



- Disclosure of Galileo IOV (Dec. 2016) and FOC (Oct. 2017) meta data by the GSA
- Disclosure of QZS-1 and QZS-2 information by JAXA in several steps in 2017

⇒ ... are very appreciated!

... triggered some of the latest model changes

- Observation biases:
 - Change from differential code biases (DCB) to observable-specific biases (OSB) => mainly internal impact (e.g., on ambiguity resolution, multi-GNSS clock solution)
- Antenna calibrations:
 - Use of disclosed antenna phase center offset (PCO) for Galileo IOV and QZS-2
 - Values included or to be included in IGS14-ANTEX file
 - Impact of ANTEX changes analyzed for Galileo IOV (=> Villiger et al. at IGS Workshop 2017)
- Attitude models:
 - Use of disclosed Galileo IOV model for all Galileo SC
- Earth albedo and transmit antenna thrust:
 - Activated for Galileo and QZS-1 (see following slides)

Experiment: Impact of satellite meta data on POD

Information available/assumed for albedo and antenna thrust experiment (data base: MGEX data of first 2 months in 2017):

- Galileo:**
- Disclosed IOV meta data (satellite mass, size, and surface properties) => sufficient for simple box-wing model
 - Later: updated solar panel properties
 - Disclosed IOV attitude model
 - Same models assumed for FOC (info meanwhile published)
 - Estimated antenna transmit power for IOV and FOC provided by Steigenberger et al. at EGU and IGS WS 2017
- QZS-1:**
- Very coarse info about QZS-1 size provided (e.g., on MGEX website); assumption on surface properties => rough guess on simple box-wing (BW) model
 - Wide range of possible SC masses (1800 - 4100 kg)
 - Transmission power provided by Kogure et al. in: Springer Handbook of Global Navigation Satellite Systems
 - Later: Updated BW model as assumed by Montenbruck et. al (2017)
 - QZS-2: not yet considered in the experiment

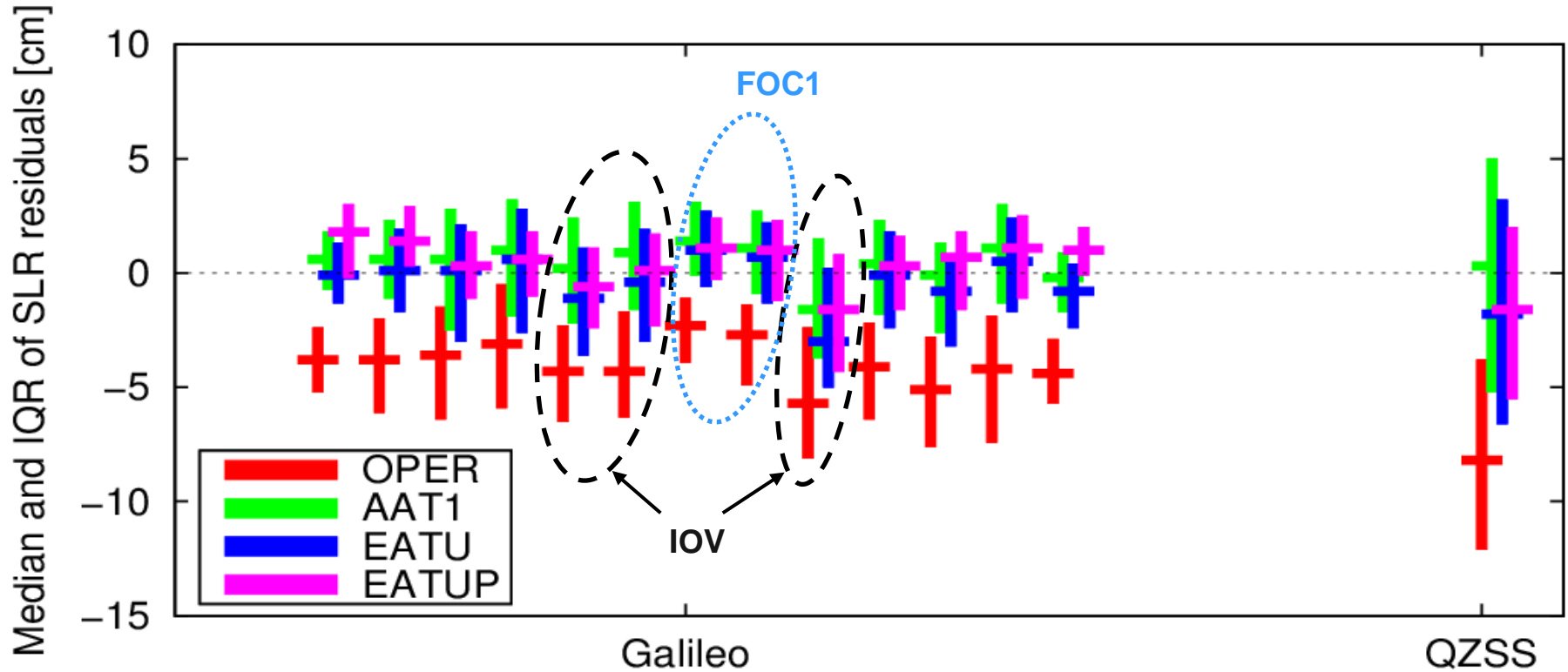
Experiment: Impact of satellite meta data on POD

Test	Galileo					QZS-1		
Name	Albedo	Ant. Thr.	Attitude	Pulses	Median SLR [cm]	Albedo	Ant. Thr. (244 W)	Median SLR [cm]
OPER	-	-	-	-	-3.8 (-+4.5)	-	-	-7.8
ALB1	x	-	-	-	-2.0	m= 1800 kg	-	-2.6
AAT1	x	260 W	-	-	+0.6	m= 1800 kg	m= 1800 kg	+0.3
AAT2	x	130 W	-	-	-0.7	m= 3600 kg	m= 3600 kg	-3.7
EAT	x	200 W	x	-	0.0	m= 1950 kg	m= 1950 kg	-0.3
EATP	x	200 W	x	R, S, W; 12h	+0.6	m= 1950 kg	m= 1950 kg	-0.3
EATU (upd...)	x	I: 130 W F:200 W	X	-	-0.2 (-+4.6)	m= 2000 kg	m= 2000 kg	-1.8
EATUP (...BW)	x	I: 130 W F:200 W	X	R, S, W; 12h	+0.5 (-+3.5)	m= 2000 kg	m= 2000 kg	-1.6 (w. PLS)

Experiment: Impact of satellite meta data on POD

Test	Galileo					QZS-1		
Name	Albedo	Ant. Thr.	Attitude	Pulses	Median SLR [cm]	Albedo	Ant. Thr. (244 W)	Median SLR [cm]
OPER	-	-	-	-	-3.8 (-+4.5)	-	-	-7.8
ALB1	x	Impact albedo: +1.8 cm			-2.0	m=1800 kg	-	-2.6
AAT1	x	260 W	-	-	+0.6	m=1800 kg	m=1800 kg	+0.3
AAT2	x	100 W	Impact antenna thrust: 1 cm/100 W		-0.7	m=1800 kg	m=1800 kg	-3.7
EAT	x	200 W	x	-	0.0	Impact SC mass: 2.2 cm/1000 kg (macro model over-scaled)		-0.3
EATP	x	200 W	x	R, S, W; 12h	+0.6	m=1800 kg	m=1800 kg	-0.3
EATU (upd...)	x	I: 130 W F: 200 W	X	-	-0.2 (-+4.6)	m=2000 kg	m=2000 kg	-1.8
EATUP (...BW)	x	F: 200 W	-	R, S, W; 12h	+0.5 (-+3.5)	m=2000 kg	m=2000 kg	-1.6 (w. PLS)

Experiment: Impact of satellite meta data on POD

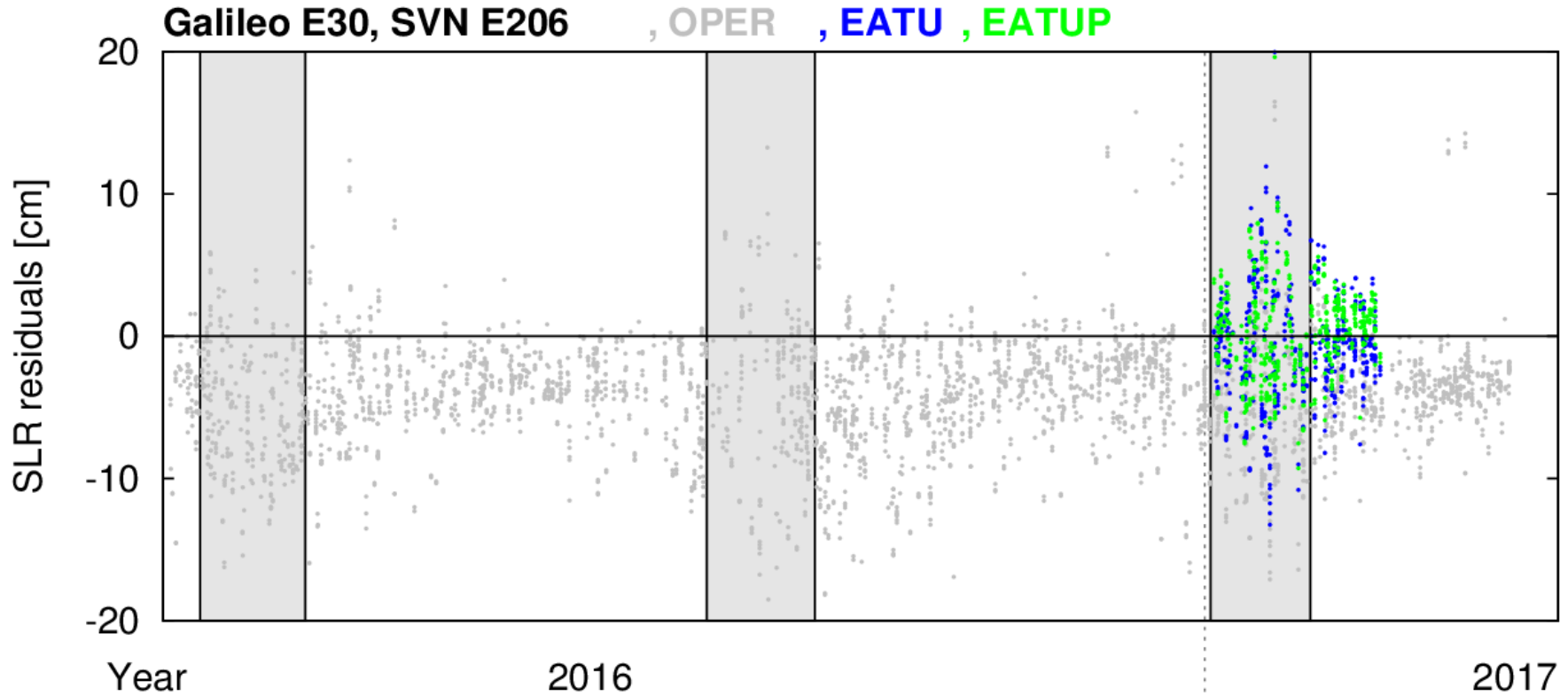


⇒ Consideration of albedo and antenna thrust reduces SLR offset

- ⇒ Uncertainties remain:
- Satellite macro model is rough and sometimes needs corrections (e.g., absorption of SP)
 - True satellite mass and CoM not always known
 - Uncertainties w.r.t. transmit power
 - Antenna calibration also impacts orbit scale

Astror

Experiment: Impact of satellite meta data on POD

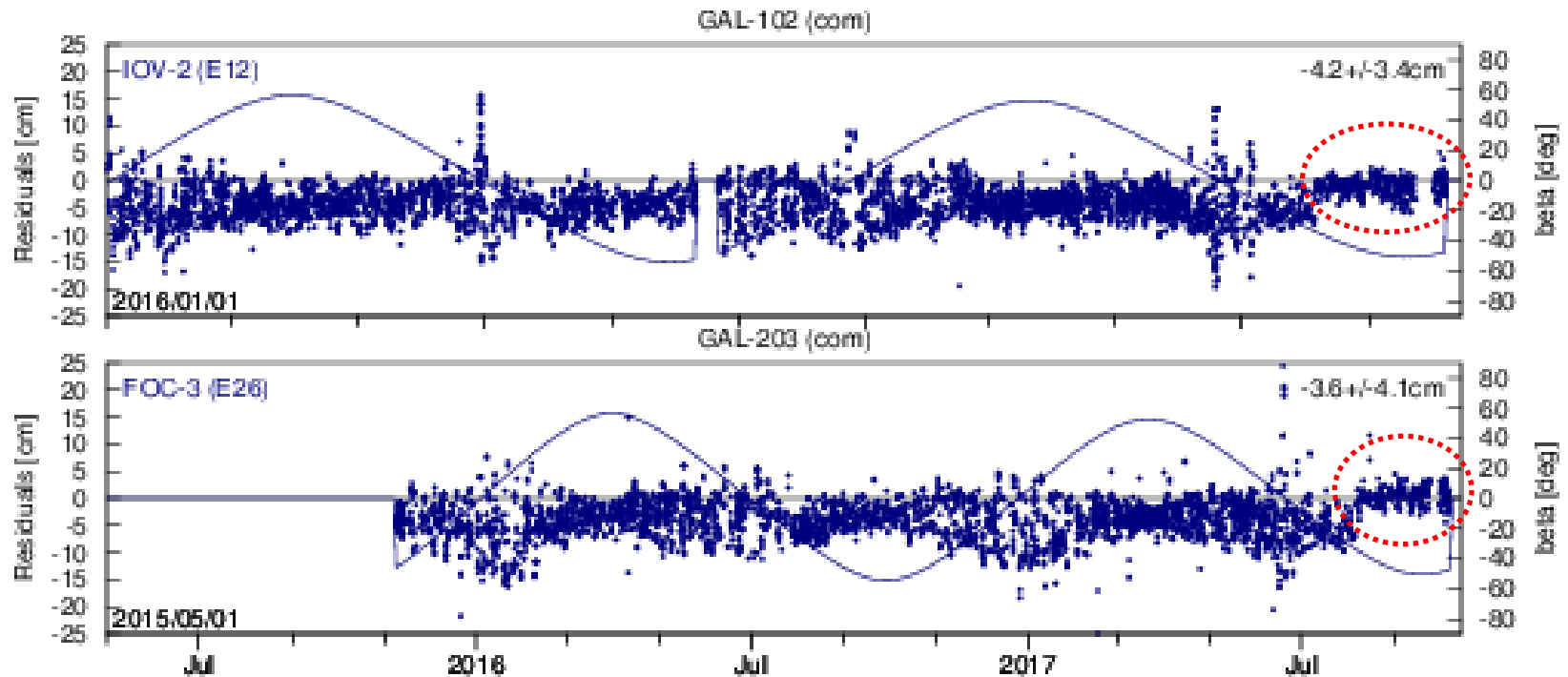


⇒ IOV attitude model (OPER vs. EATU): slight impr. during deep eclipse

⇒ Pulses in R, S, W (EATUP): IQR drops 4.6 → 3.5 cm
(expected future configuration)

⇒ Orbit errors remain increased during eclipses (unmodelled thermal re-radiation from spacecraft?)

Experiment: Impact of satellite meta data on POD



(SLR validation provided by the IGS MGEX: <http://mgex.igs.org>)

- ⇒ Model changes active in CODE MGEX solution since GPSWEEK 1962
- ⇒ Expected orbit improvement confirmed by external validation

Experiment: Effect of IOV antenna calibrations

Table: Average inter-system bias for IOV satellites (E11,E12,E19) in cm

Sol	East	North	Up	σ_{East}	σ_{North}	σ_{Up}
IGS	-2.19	-0.15	2.27	32.02	16.56	33.10
IOV	-0.96	-0.23	1.70	22.54	11.47	19.69

Consistency between GPS and Galileo improves (data base: IGS MGEX network)

STA	Antenna type <i>Used PCO and PCV:</i>	GTRA			
		L1/L2 [mm]	L1/L5 [mm]	Δ GTRA [mm]	Δ PCO [mm]
BRUX	JAVRINGANT_DM NONE	3.7	15.0	-11.3	-8.56
POTS	JAV_RINGANT_G3T NONE	4.9	15.9	-11.0	-9.22
...
ISTA	LEIAR25.R4 LEIT	6.5	14.6	-8.1	-6.53

Remaining limiting factor: Adopting receiver antenna PCO+PCV from L1/L2 frequencies, might cause coordinate offsets with magnitude of PCO difference (data base: EUREF)

(source: Villiger et al. (2017): Consistency of antenna products in the MGEX environment. IGS Workshop 2017)

⇒ Impact of satellite antenna calibrations depends also on availability of corresponding ground antenna calibrations for all GNSS and frequencies (=> IGS task)

CODE MGEX: next steps

Galileo IOV: - Use of transmit antenna PCV calibrations after inclusion in IGS ANTEX file (released on 23 Oct. 2017)

Galileo FOC: - Update of box-wing model for albedo
- Use of antenna PCO and PCV calibrations (investigation of impact planned)
(meta data available since Oct. 2017) - Update of attitude model
- Update of antenna transmit power (as soon as data is disclosed)

QZS-1: - Update of box-wing model for albedo
(new info in Oct. 2017) - Update of transmit power and satellite mass
- Normal attitude and related SRP model

QZS-2: - Use of antenna PCO and PCV calibrations
(basic set of info provided in June 2017) - Set up box-wing model for albedo modelling and antenna thrust as soon as all necessary data is provided

Summary and conclusions

- Lack of meta information about spacecraft properties results in poor performance of «precise» orbit determination (POD) - especially for new GNSS
- This limits the usefulness and acceptance of these GNSS for high precision geodetic applications (e.g., ITRF scale)
- Some system providers (namely GSA, JAXA) have recognized this and started to release meta data within the recent months
- Besides information gathered by the scientific community itself (e.g., antenna PCO estimations, antenna transmit power estimations) this information is starting to improve GNSS solutions
- The provision of meta data is therefore appreciated; release of further and more complete satellite meta data is encouraged

**Thank you
for
your attention!**