Simulation of realistic SLR observations to optimize tracking scenarios

F. Andritsch, A. Maier, R. Dach, A. Jäggi

Astronomical Institute, University of Bern, Switzerland

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Introduction: Overview

- scenarios Germany to optimize tracking s - 2016, GFZ Potsdam, October observations Ranging, 10 Simulation of realistic SLR Florian Andritsch: Simulation of realis 20th International Workshop on Laser
- Simulation of Satellite Laser Ranging measurements to GNSS and geodetic satellites.
- Calculation of pseudorange due to geometry, then apply:
 - Corrections
 - Noise
 - Synthetic observations upon which comparison and optimization can be done.

Impact of target selection.

Introduction: Procedure

Bernese GNSS Software.

- Independent noise generation for each observation.
 - Adding/removing satellites at other epochs.
 - Exchanging satellites at specific epoch.
- Observation selection separate from simulation.



Introduction- ILRS Tracking Campaigns

Campaign1: August 01 - September 30, 2014

- All GNSS satellites (on ILRS priority list, 18 satellites); more if able
- Three sets of two normal points distributed over transit; normal point includes 1000 FR points or last 5 minutes, whichever is shorter

Campaign2: November 22,2014- February 28, 2015

- Six GLONASS as first priority, Beidou and Galileo as second priority, remaining GLONASS as third priority
- minimum three segments along each pass with three NPTs in each segment

Campaign3: August 20 - October 16, 2015

- Six GLONASS as first priority, Compass-M3 and Galileo as second priority, remaining GLONASS as third priority
- Nine NPTs over the pass; 3 during the ascending/early region, 3 in the central region, 3 in the descending/late region of the pass



Introduction- ILRS Tracking Campaigns

Common Results

- Need more data
- Few stations could fulfill requirements all the time
- More daylight data

ILRS can handle tracking of all the required satellits (for now).

Simulation might give a definite answer in future.



Simulation-Requirements

- scenarios Germany tracking s Potsdam, (016, GFZ optimiz \sim 0 observations to Florian Andritsch: Simulation of realistic SLR obser 20th International Workshop on Laser Ranging, 10
- Ability to generate synthetic SLR measurements to satellites in form of NP.
- Include station/satellite specific noise handling.
- Based on final orbit products (or predictions).
- Possibility to alter observations as needed.
- Lie within the accuracy requirements of the ILRS.



Simulation – Implementation

- scenarios I, Germany tracking s Potsdam, (0 optimize , to Simulation of realistic SLR observations t Workshop on Laser Ranging, 10 October Florian Andritsch: 20th International
- Separate SLR mode within Bernese GNSS Software Simulation tool.
 - Selectable noise
 - Selectable stations
 - Selectable observation list
- Result are range observation files.



Simulation - Observation Selection

YYYY	HH	DD	STAT	TIME IN SECONDS	PRN	
****	**	××	****	************	***	×
2015	05	06	7810	32046.185902247878	108	1
2015	05	06	7810	32126.060302248032	108	1
2015	05	06	7810	34742.648702320352	107	1
2015	05	06	7810	34862,552302311269	107	1
2015	05	06	7810	35154.578702318162	107	1
2015	05	06	7810	36243.455102316861	108	1
2015	05	06	7810	36376,229502318871	108	1
2015	05	06	7810	40050,113102219737	111	1
2015	05	06	7810	50219.056302341145	111	1
2015	05	06	7810	51928.352112342225	121	1
2015	05	06	7810	51932,583932335045	121	1
2015	05	06	7810	51936.500912335854	121	1
2015	05	06	7810	51942,945872333563	121	1
2015	05	06	7810	51946.872112332734	121	1
2015	05	06	7810	51951,641012342428	121	1
2015	05	06	7810	51958,623052340256	121	1
2015	05	06	7810	51962,188152332608	121	1
2015	05	06	7810	51967.808972341074	121	1
2015	05	06	781 0	51972.374152341465	121	1
2015	05	06	7810	51977.856072334798	121	1
2015	05	06	781 0	51982,106412339985	121	1
2015	05	06	7810	51987,792052336998	121	1
2015	05	06	7810	51992.042392342613	121	1
2015	05	06	7810	51997.524312334826	121	1
2015	05	06	7810	52001.691312335461	121	1
2015	05	06	7810	52007.451032332443	121	1
2015	05	06	7810	52018,155592334660	121	1
2015	05	06	7810	52022.054052340201	121	1
2015	05	06	7810	52027.526712332408	121	1
2015	05	06	7810	52031,951052338896	121	1
2015	05	06	7810	52037.587572334320	121	1
2015	05	06	7810	52042.560972338986	121	1
2015	05	06	7810	52047,119922336395	121	1
2015	05	06	7810	52052,738022335390	121	1
2015	05	06	7810	52056.818052335875	121	1
2015	05	06	7810	52061.561202337041	121	1
2015	05	06	7810	54224.806/0223/183	123	1
2015	05	UD OC	7810	61/00.2330822/0584	124	1
2015	05	UD OC	7810	62868,262582268966	115	1
2015	05	06	7810	63549,4331022691//	113	1
2015	05	06	7810	63843.31/90226/241	113	1
2015	05	06	7810	55903,531902260132	113	1
2015	05	06	7810	78878,631902225781	103	1
2015	05	06	7810	79022,709502223472	103	1
2015	05	06	7810	75281,855502518562	102	1
2010	00	UD D	7810	73206,484302322424	102	1

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Simulation - Noise

- White noise
 - Selectable
 - Elevation dependent sigmas
 - Repeatable
- Normal distributed noise
 - Different parameters for each station/satelite combination.
 - Resembling bin RMS found in NP files.

Simulation - First Results

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 - **First Results**
 - Without noise
 - White noise
 - Normal distributed noise

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Simulated observation file

GPSSIM2	:	simul	ate SLR data	to GNSS	satellites	18433				
MEASUREME Reference	NT TYPE: EPOCH :	RANGE 2015-	05-06 0:37:	13 (126)	CREATED : MODIFIED:	26-MAY-16 26-May-16	15:37 15:37			
# DIFFERE # FREQUEN # SATELLI # EPOCHS # FLAGGED	NCES CIES TES EPOCHS	0 1 0 77051 0	FI SI SI RI RI	DRMAT NUI ESSION I JBSESSIO 35. INTEL EMARK NUI	HBER : DENTIFIER : N IDENTIF. RYAL (S) : HBER :	6 1265 5 1.0000 0	000			
STATION N OPERATOR RECEIVER ANTENNA T RECEIVER/	AME : NAME : Type : Ype : Antenna;	ZIMM SIMUL SIMUL	14001S007 A DEFAULT A N 0 / 0	DNE						
CLOCK COR	RECTION:	POLYN	OMIAL DEG 0							
POS.ECCEN	TR. (H):	0,00	00 0.0000	0000		.		.		
SAT #L	1–OBS OK	#L1-0	BS BAD #L2-()BS OK	#L2-OBS BAD	Statio Obstype1	on1 Obstype2	Stati Obstype1	on2 Obstype2	
L1,L2 OBS OBS.N 1	ERVATIONS TIME 8:54:06	: F#S 1	RANGE (M) 21468293.62	FFS SA 5 0 108	15-05-06	A 0.000000000	T THE END: -0.000000	DATE, FR 000	ACT.(S), CLOCK	(S)
121	8:55:26	1	21555114,39	3 0 108	15-05-06	0.000000000	-0.000000	000		
413	9:39:02	1	21771033.65	7 0 107	15-05-06	0.000000000	-0.000000	000		
903	9:41:02	1	19687406.54	4 0 107	15-05-06	0.000000000	-0,000000	000		
983	9:45:54	1	19660905,18	3 0 107	15-05-06	0.000000000	-0.000000	000		
5100	10:04:03	1	19554801.69	2 0 108	15-05-06	0.000000000	-0.000000	000		
5233	10:06:16	1	19595476.023	3 0 108	15-05-06	0.000000000	-0.000000	000		
19701	11:07:30	1	19678965,28	4 0 111	15-05-06	0.000000000	-0.000000	000		
26550	13:56:59	1	21223037.74	6 0 111	15-05-06	0.000000000	-0.000000	000		
26555	14:25:28	1	21225342.65	l 0 121	15-05-06	0.000000000	-0.000000	000		
26562	14:25:32	1	21228575.44	5 0 121	15-05-06	0.000000000	-0.000000	000		
26576	14:25:36	1	21235060,79	0 121	15-05-06	0.000000000	-0.000000	000		
26582	14:25:42	1	21237848,203	3 0 121	15-05-06	0.000000000	-0.000000	000		
26586	14:25:46	1	21239709.43	6 0 121	15-05-06	0.000000000	-0.000000	000		

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White noise





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Normal distributed noise



Noise parameters



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Noise parameters



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Conclusions & Outlook

Promising results that compare well in terms of residuals for existing observations.

- Different tracking strategies will be generated and used for comparison
- Investigating the impact of reducing observations to specific satellites in favor of more observations to others

Thank you for your attention.



References

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