

# Orbit modelling in CODE's MGEX solution

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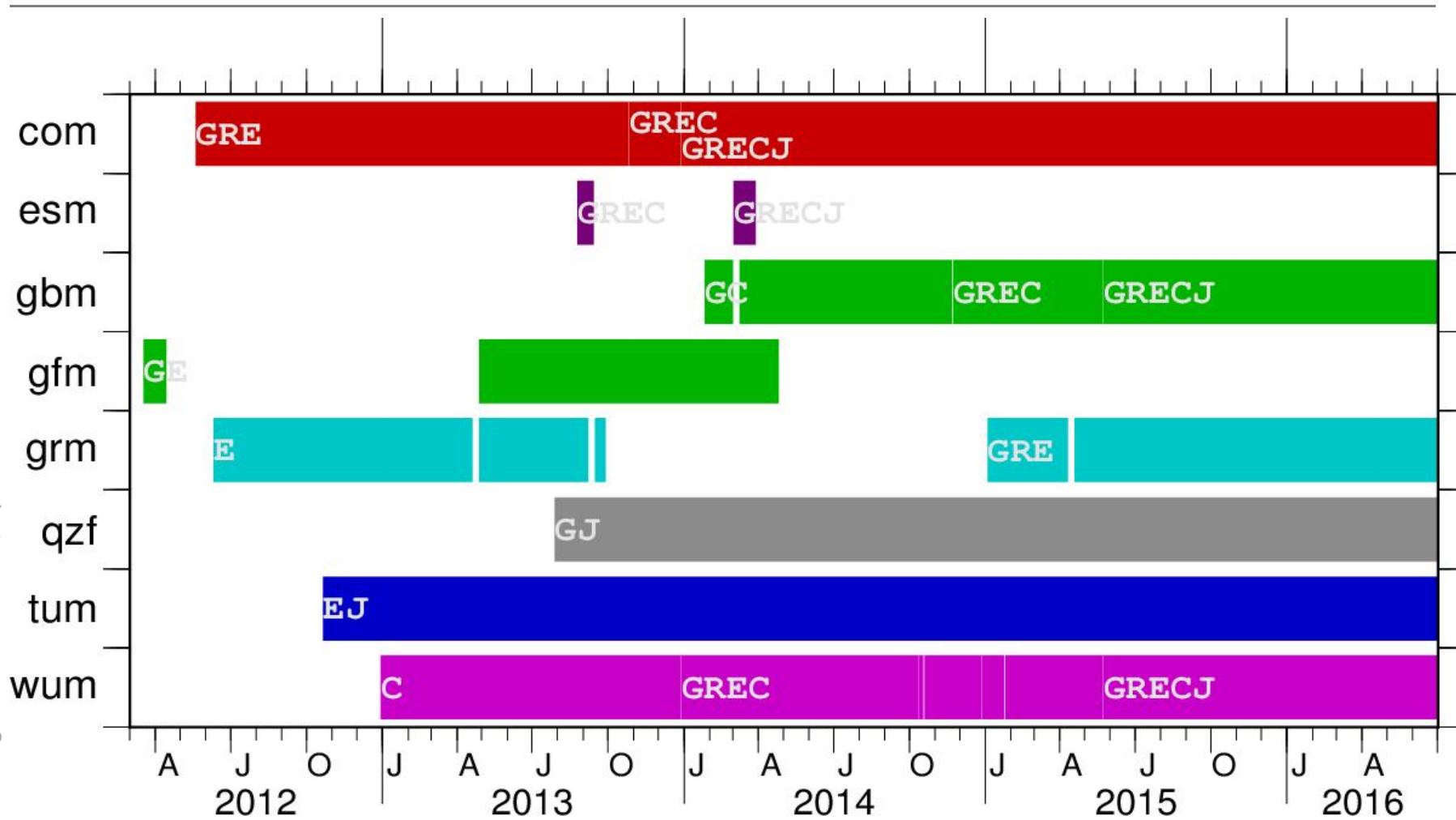
ICG+2016, 27-30 July 2016,  
Shanghai, China

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- Data base and network
- CODE MGEX orbit solution
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- Validation of satellite orbits and clock corrections
- Summary and outlook

# MGEX products availability



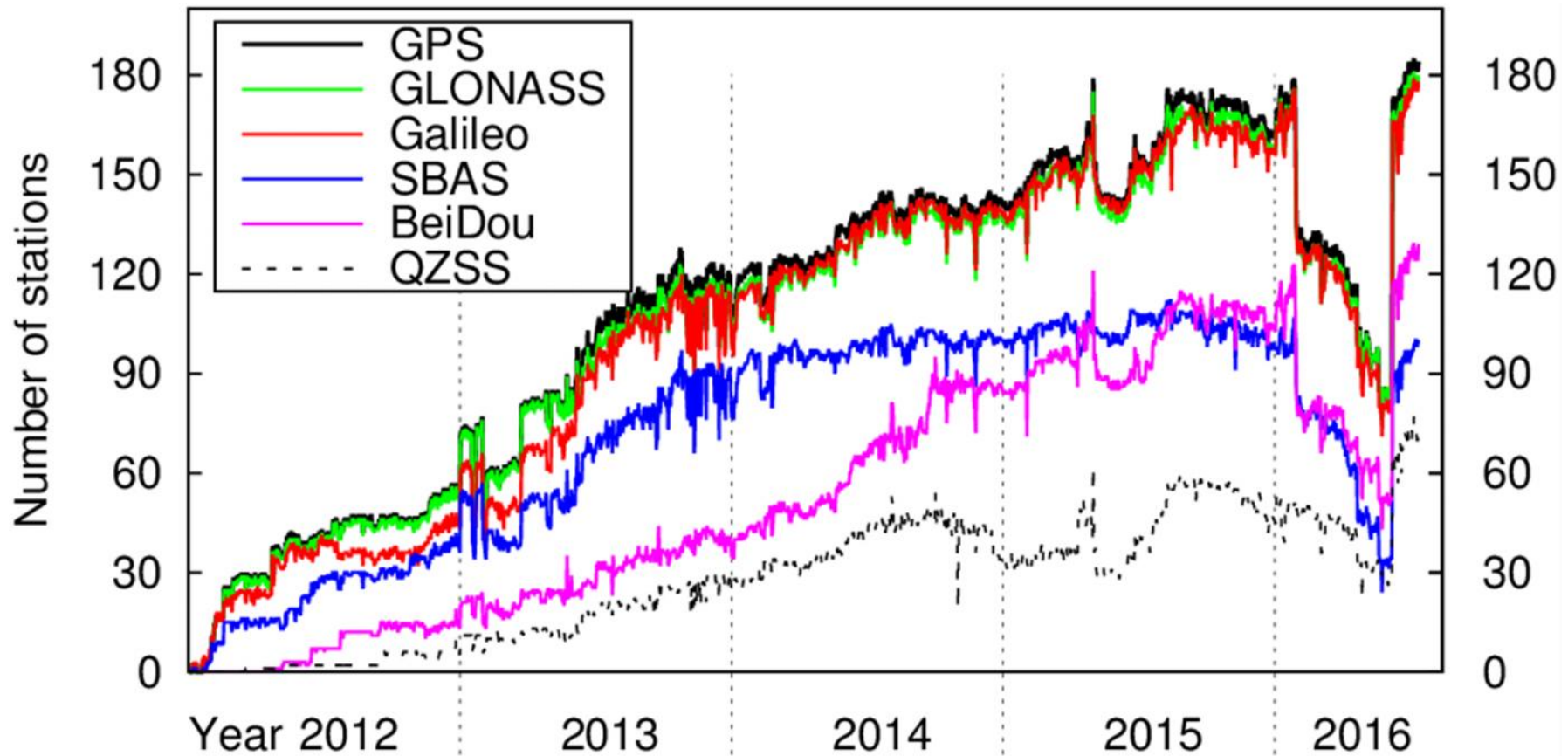
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Status: July 2016

Satellite system IDs according to the content of the precise orbit files at <ftp://cddis.gsfc.nasa.gov/pub/gps/products/mgex/>

# MGEX data monitoring

Number of stations providing daily RINEX3 files and included in CODE's raw data monitoring (data sources: IGS and EPN)



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# CODE MGEX orbit solution

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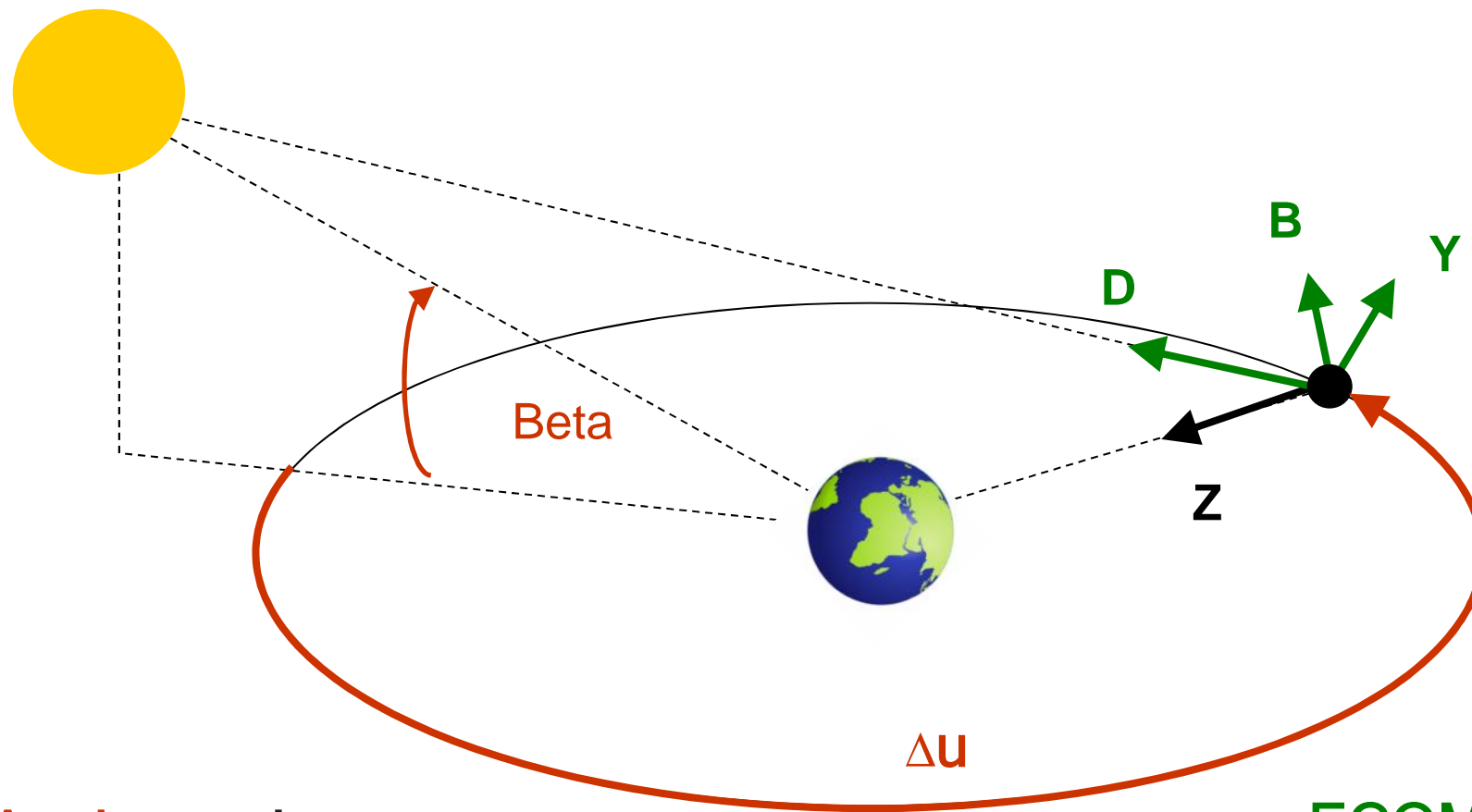
GNSS considered:	GPS + GLONASS + Galileo + BeiDou (MEO+IGSO) + QZSS (≈70 SV)
Processing mode:	Post-processing (≈2 weeks latency)
Timespan covered:	GPS-weeks 1689 - today
Number of stations:	130 (GPS), 110 (GLONASS), 85 (Galileo); 55 (BeiDou); 20 (QZSS)
Processing scheme:	Double-difference network processing (observable: phase double differences)
Signal frequencies:	L1+ L2 (GPS + GLO+ QZSS); E1 (L1) + E5a (L5) Galileo; B1 (L1) + B2 (L7) BeiDou
Orbit characteristic:	3-day long arcs; SRP: ECOM / ECOM2 (since 2015)
Reference frame:	IGS08 (until week 1708); IGb08 (since week 1709)
IERS conventions:	IERS2003 (until 1705); IERS2010 (since 1706)
Product list:	Daily orbits (SP3) and ERPs
Distribution:	<a href="ftp://cddis.gsfc.nasa.gov/gnss/products/mgex/">ftp://cddis.gsfc.nasa.gov/gnss/products/mgex/</a> and <a href="ftp://ftp.unibe.ch/aiub/CODE_MGEX/">ftp://ftp.unibe.ch/aiub/CODE_MGEX/</a>
Designator:	comwwwwd.???.Z

# CODE MGEX clock solution

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GNSS considered:	GPS + GLONASS + Galileo + BeiDou + QZSS ( $\approx 70$ SV)
Processing mode:	Post-processing ( $\approx 2$ weeks latency)
Timespan covered:	GPS-weeks 1710 - today
Number of stations:	130 (GPS), 45 (GLO), 45 (Galileo); 50 (BeiDou); 20 (QZSS)
Processing scheme:	<b>Zero-difference</b> processing (observable: code+phase undifferenced)
Signal frequencies:	L1+ L2 (GPS + GLO+ QZSS); E1 (L1) + E5a (L5) Galileo; B1 (L1) + B2 (L7) BeiDou
A priori information:	Orbits, ERPs, coordinates, and troposphere from CODE MGEX orbit solution introduced as known
Reference frame:	IGb08
IERS conventions:	IERS2010
Product list:	Epoch-wise (300s) satellite and station clock corrections in daily clock RINEX files; daily inter-system biases for mixed stations in Bernese DCB and BIAS-SINEX format <a href="ftp://cddis.gsfc.nasa.gov/gnss/products/mgex/">ftp://cddis.gsfc.nasa.gov/gnss/products/mgex/</a> and <a href="ftp://ftp.unibe.ch/aiub/CODE_MGEX/">ftp://ftp.unibe.ch/aiub/CODE_MGEX/</a>
Distribution:	

# Orbit description and Yaw attitude



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## Angles and vectors:

**Beta:** Elevation of Sun above orbital plane

**$\Delta u$ :** Argument of latitude

**Z:** Direction satellite  $\rightarrow$  Earth (antenna direction)

## ECOM axes:

**D:** Direction satellite  $\rightarrow$  Sun

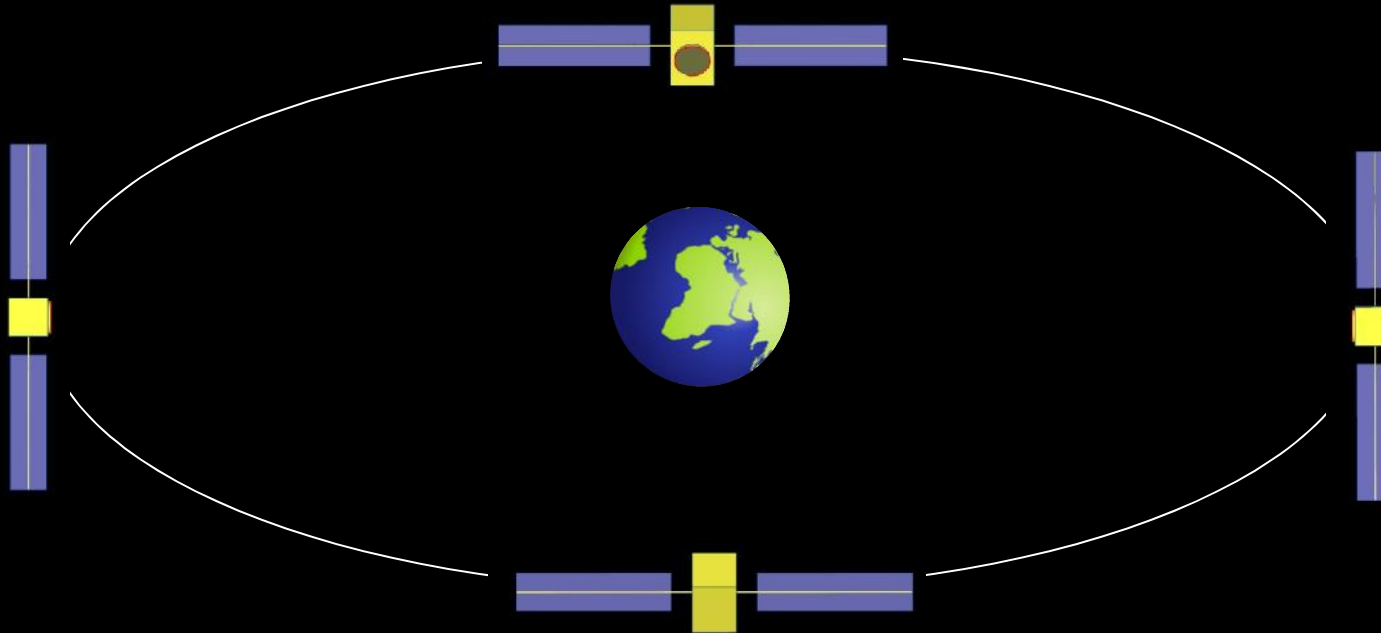
**Y:** Solar panel axis

**B:** Third ECOM axis

# Solar radiation pressure

Satellite cross-section as seen from the Sun (Beta  $\approx 30^\circ$ ) during one orbital revolution:

=> solar panel area does not change



=> but: cross-section of long satellite bodies w.r.t. the Sun varies



# New Empirical CODE radiation pressure Model

- MGEX-reprocessing for 2014 using ECOM (5 RPR par.; Beutler et al., 1994, Springer et al., 1999) vs. ECOM2 (9 RPR par., Arnold et al., 2015)
  - Validation with SLR residuals and satellite clock corrections
  - The new ECOM takes into account the periodically changing cross section of elongated satellite bodies wrt. the Sun
- => Improvements expected for Galileo, GLONASS, QZSS

## ECOM1 (old):

$$D(u) = D_0$$

$$Y(u) = Y_0$$

$$B(u) = B_0 + B_C \cos(u) + B_S \sin(u)$$

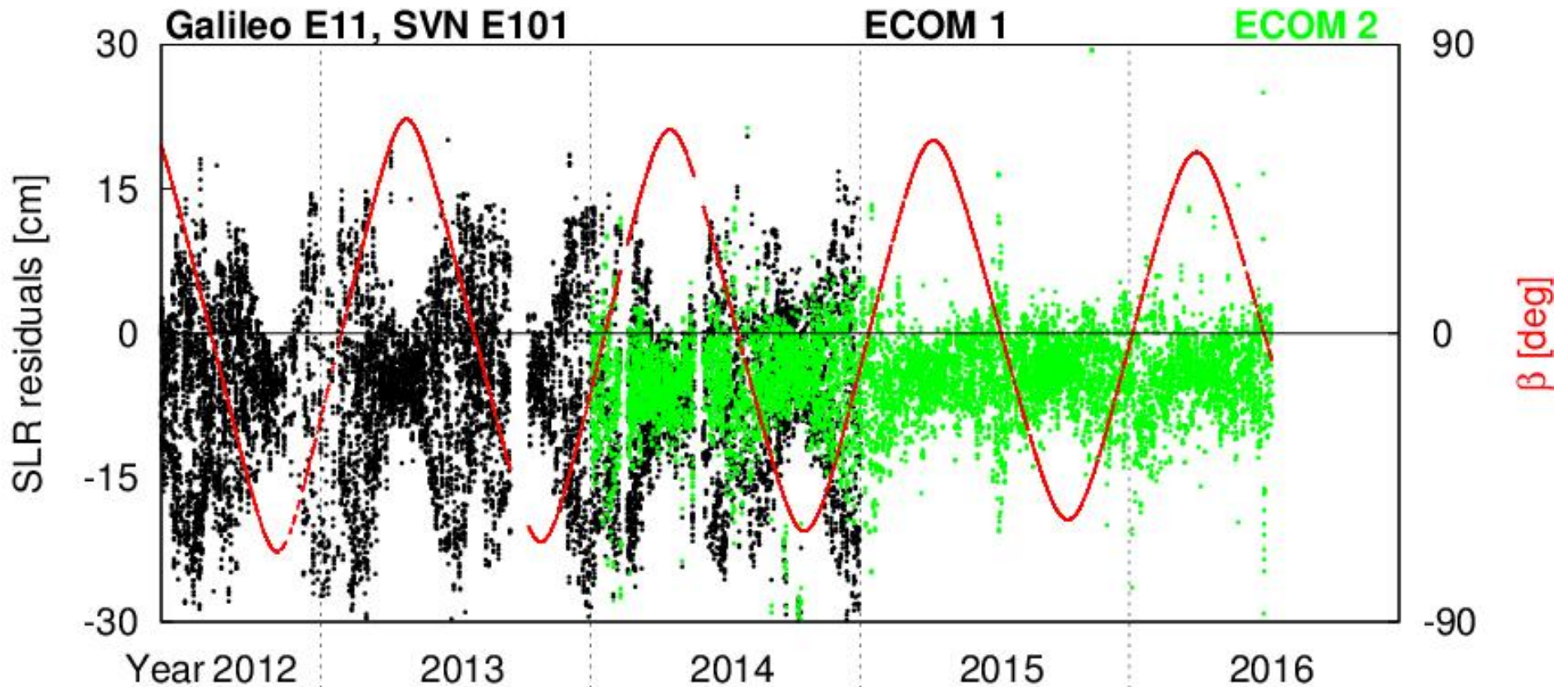
## ECOM2 (new):

$$D(u) = D_0 + D_{2C} \cos(2\Delta u) + D_{2S} \sin(2\Delta u) \\ + D_{4C} \cos(4\Delta u) + D_{4S} \sin(4\Delta u)$$

$$Y(u) = Y_0$$

$$B(u) = B_0 + B_C \cos(\Delta u) + B_S \sin(\Delta u)$$

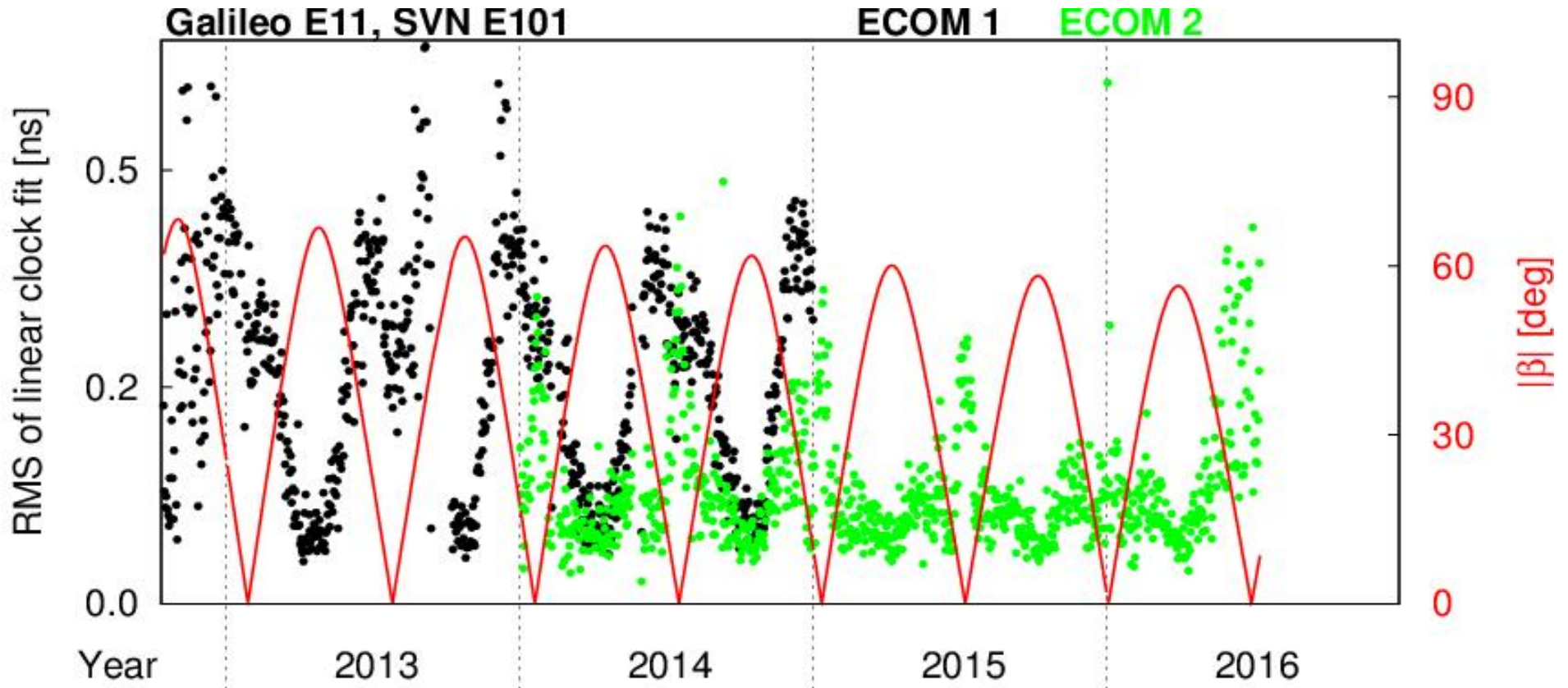
# Galileo orbit validation



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⇒ Significant reduction of dependency on beta-angle, when changing to the ECOM2

# Galileo clock validation

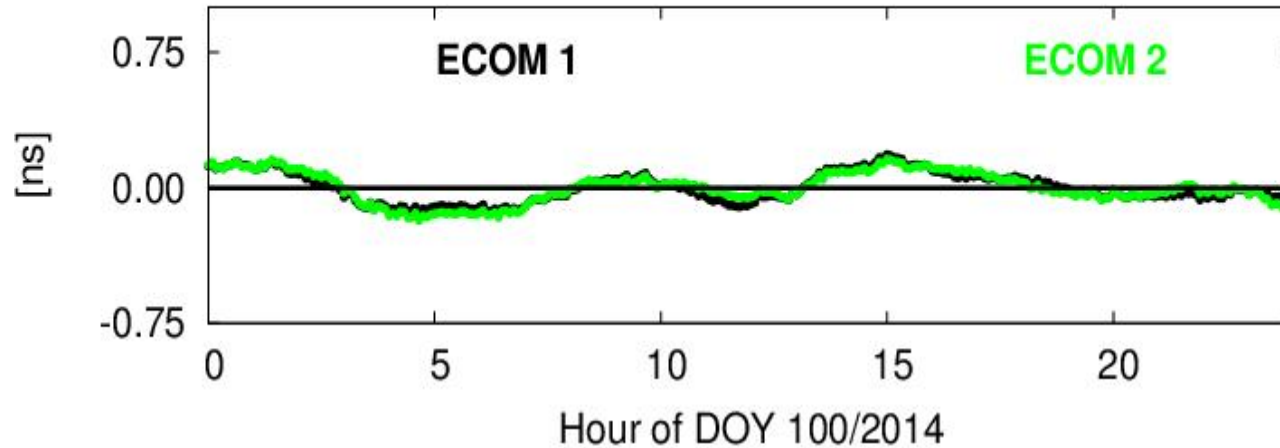


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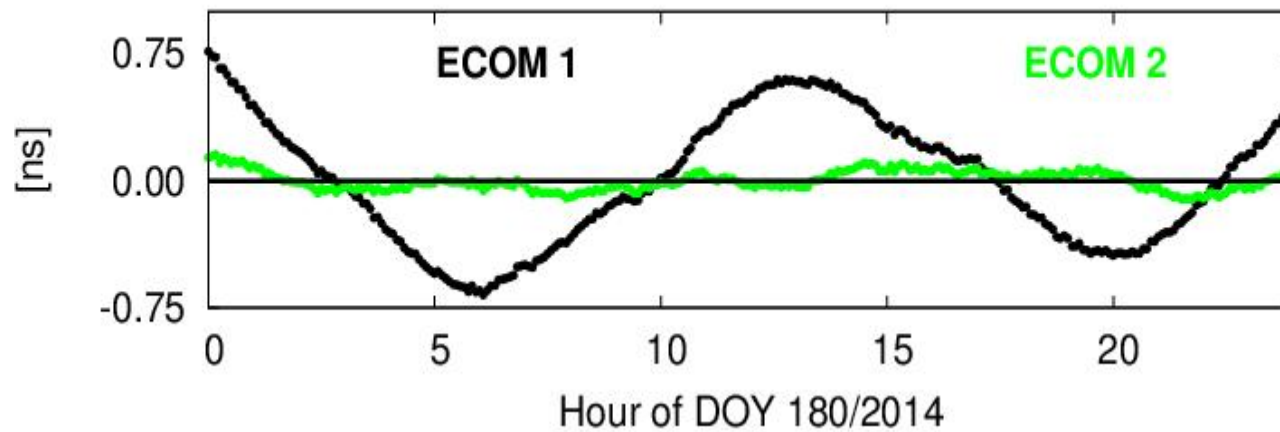
- ⇒ Significant reduction of dependency on beta angle
- ⇒ Pronounced signal remains during eclipse season or close-by

# Galileo clock validation

Clock corrections of Galileo E11, SVN E101



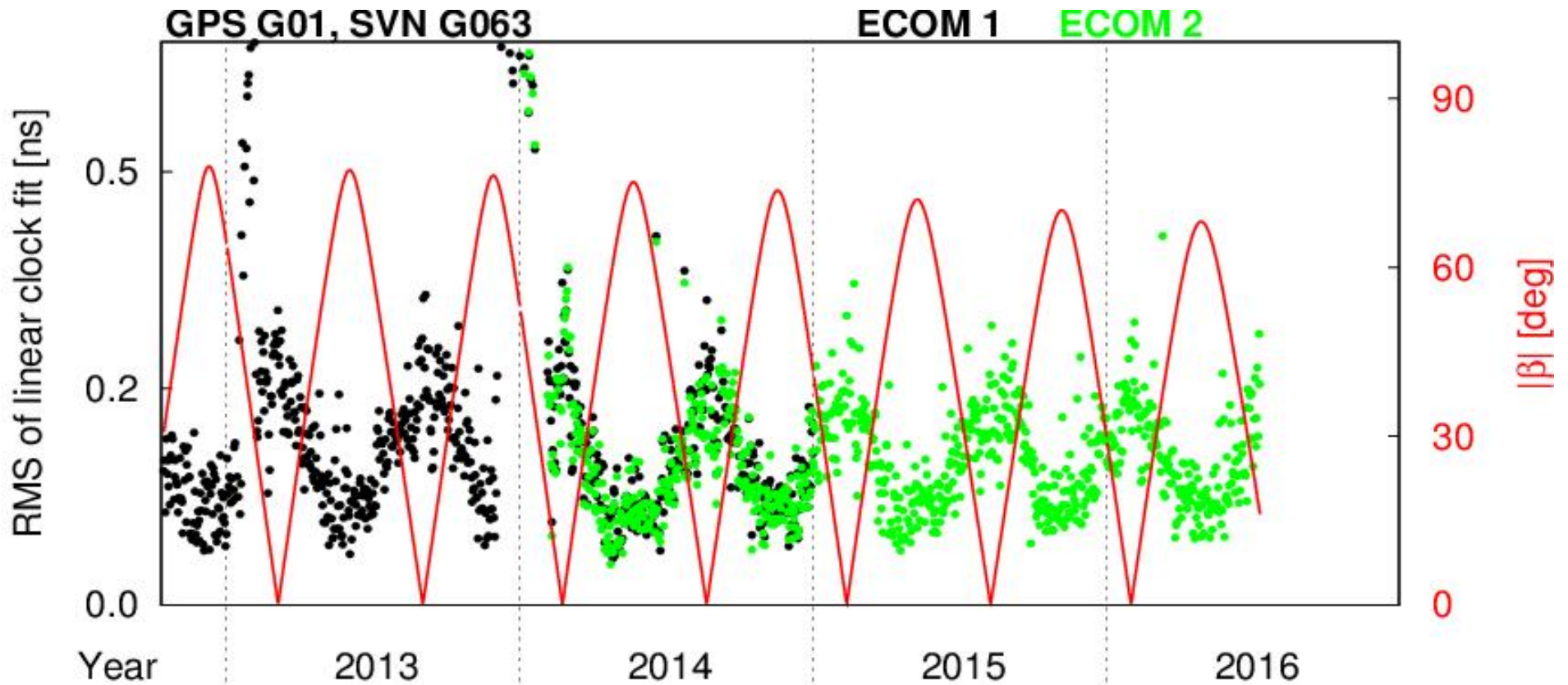
Large beta-angle:  
=> Clock signal has small amplitude (about  $\pm 0.15$  ns)



Small beta-angle:  
=> Periodic signal caused by mis-modelled orbit (ECOM1)

=> Significant reduction of signal amplitude from  $\pm 0.75$  ns to  $\pm 0.15$  ns when switching to **ECOM2**

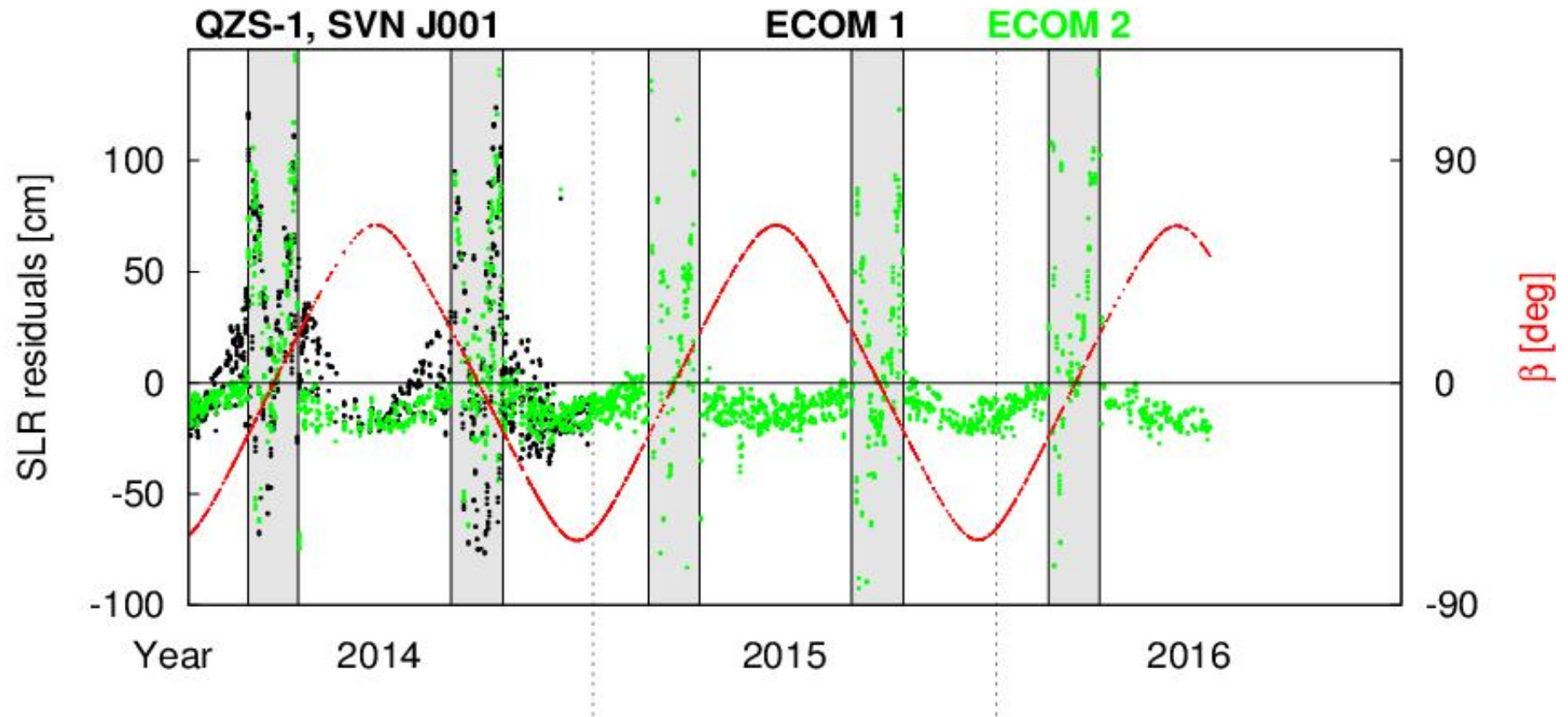
# GPS clock validation



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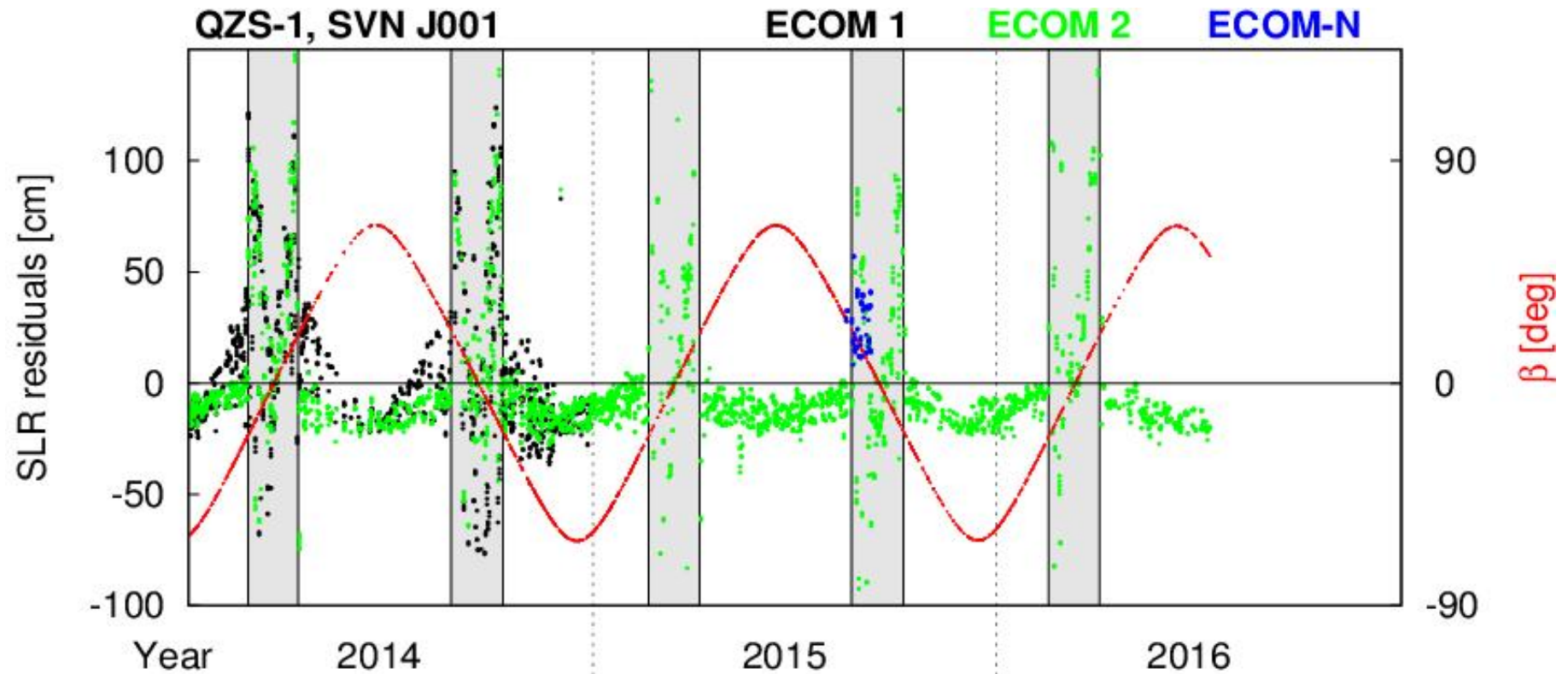
- ⇒ Moderate dependency on beta angle
- ⇒ No impact of new ECOM
- ⇒ Different possible reasons for signal (e.g., thermal issues)

# QZSS orbit validation



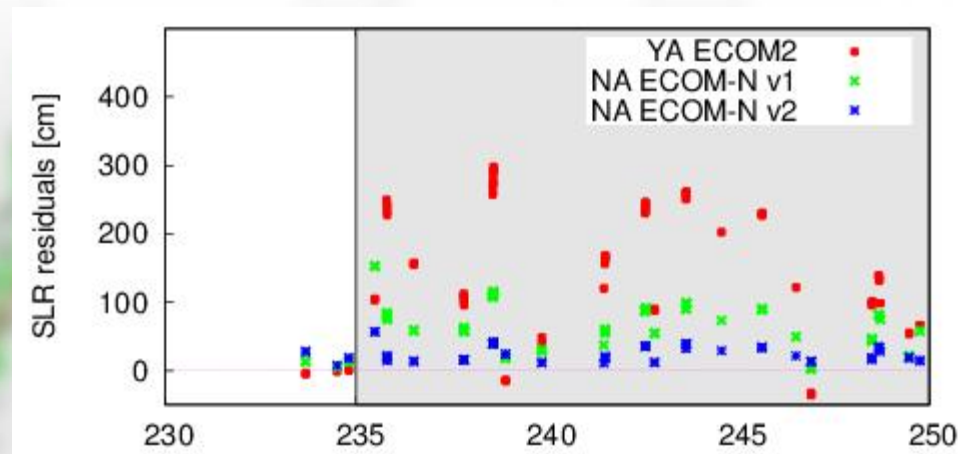
- ⇒ Yaw-steering: ECOM2 reduces dependency on beta angle
- ⇒ Significant SLR offset remains

# QZSS orbit validation



- ⇒ Normal attitude mode ( $|\beta| < 20^\circ$ ; marked grey): large orbit errors
- ⇒ Test of new ECOM versions better suited for orbit normal attitude mode

# QZSS orbit validation

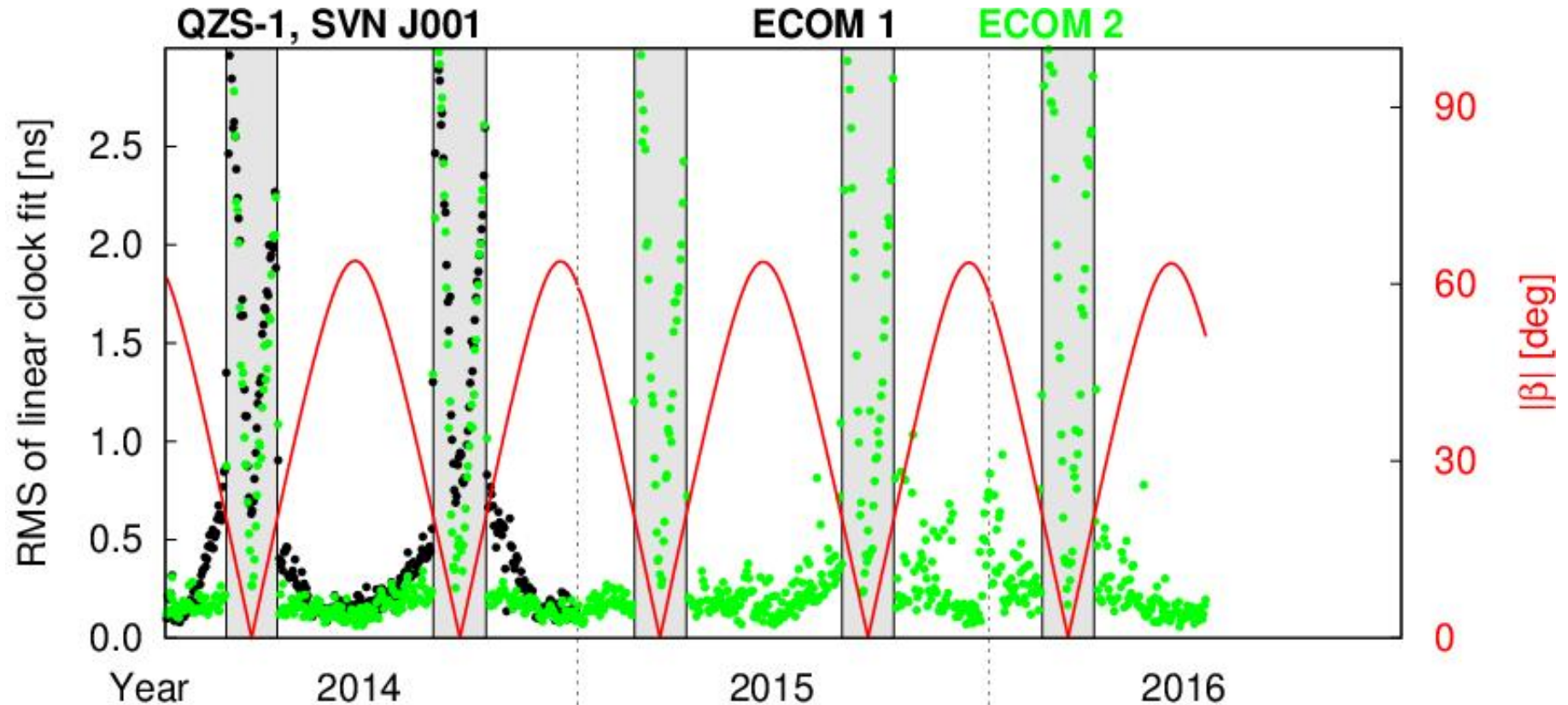


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⇒ Test of new ECOM versions better suited for orbit normal attitude mode

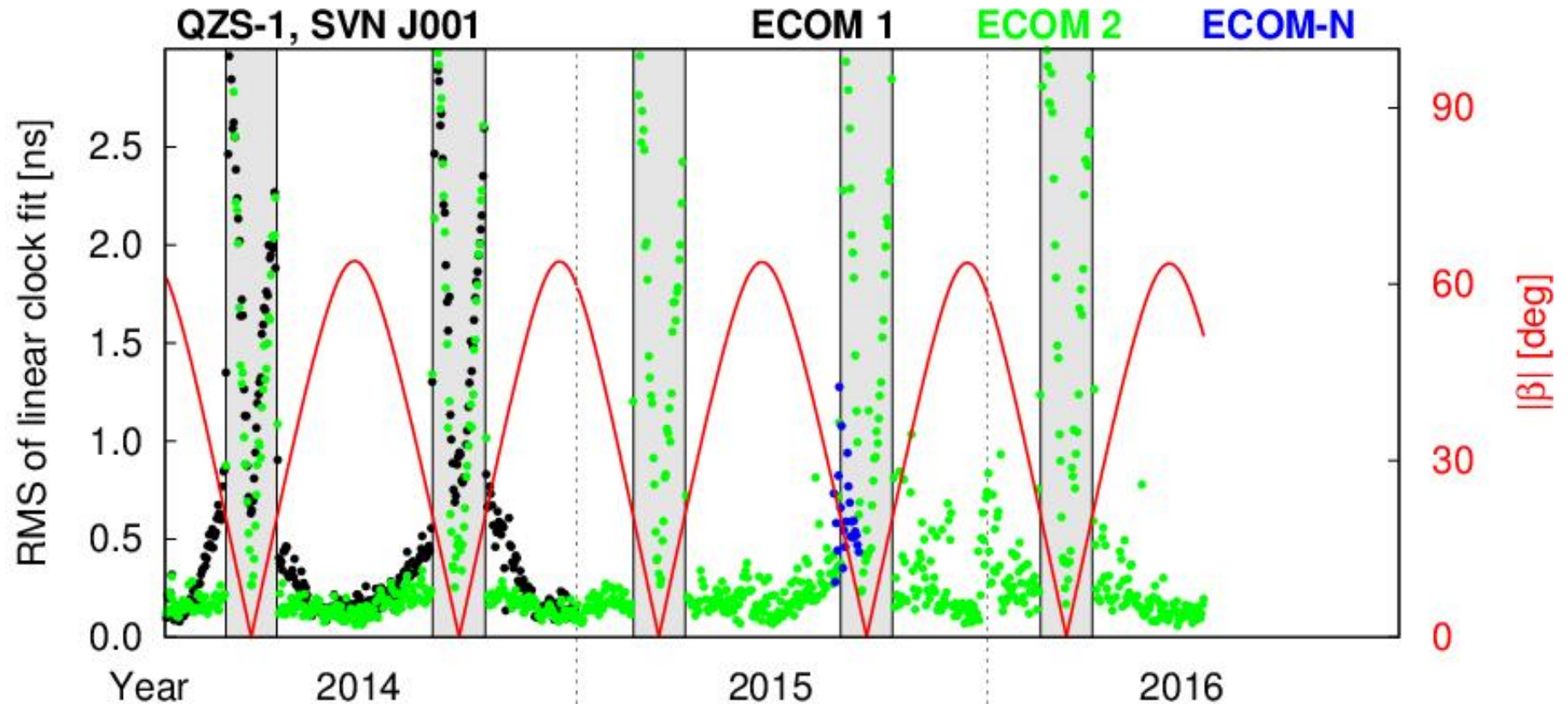


# QZSS clock validation



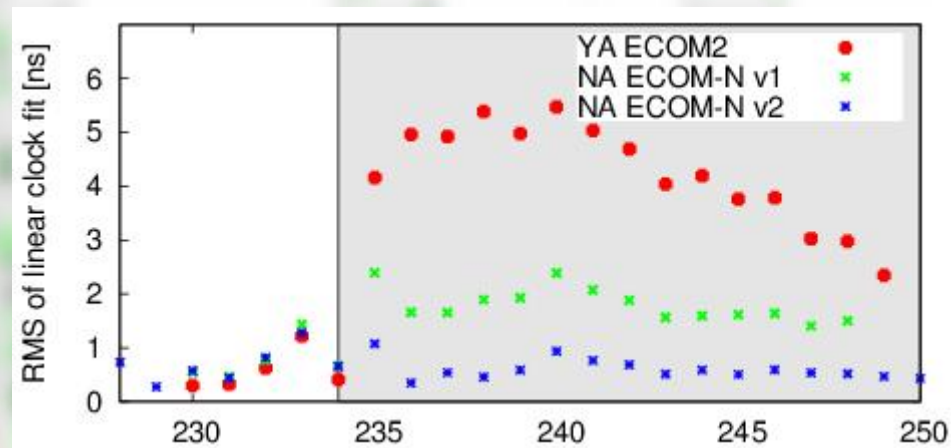
- ⇒ Yaw-steering: significant reduction of dependency on beta angle thanks to ECOM2
- ⇒ Orbit normal attitude mode (grey): large errors remain

# QZSS clock validation



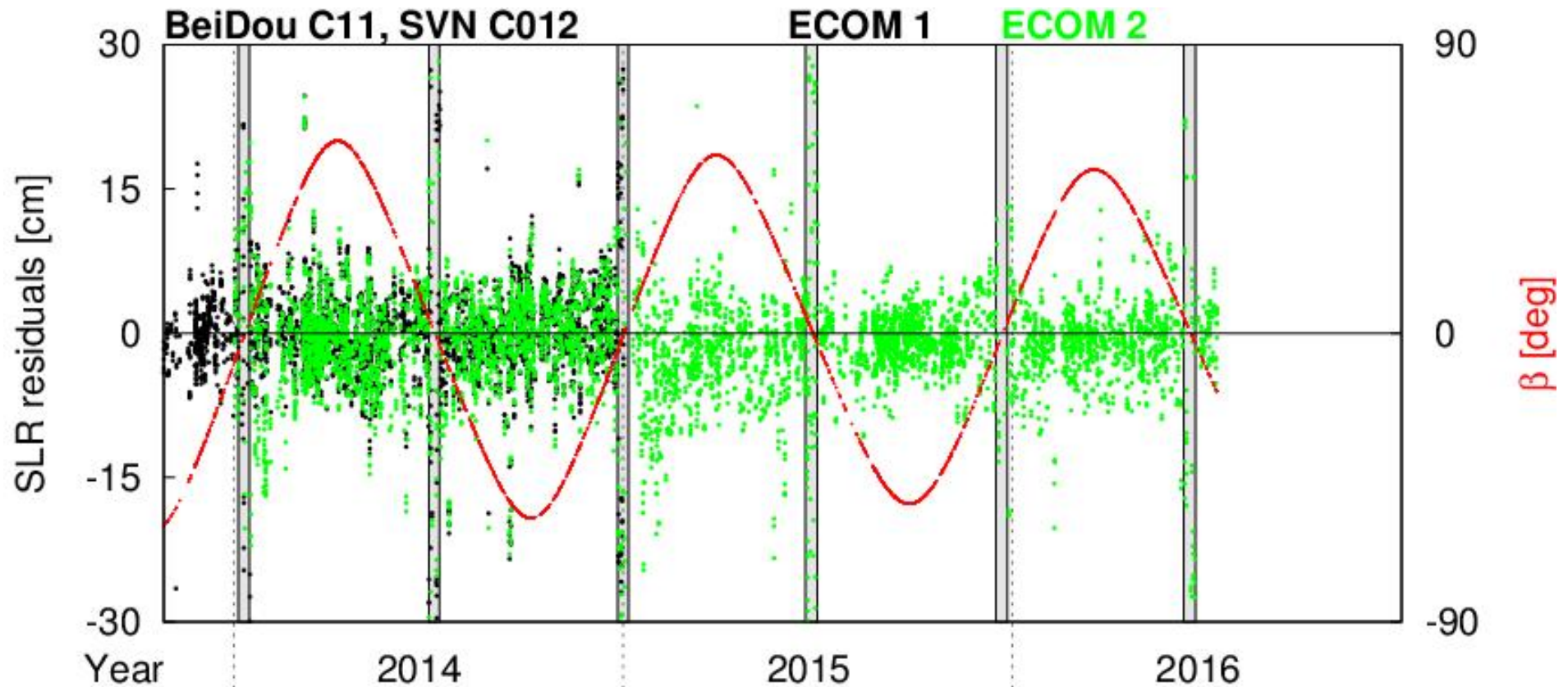
⇒ Experiments with ECOM versions better suited for orbit normal attitude mode

# QZSS clock validation



⇒ Experiments with ECOM versions better suited for orbit normal attitude mode

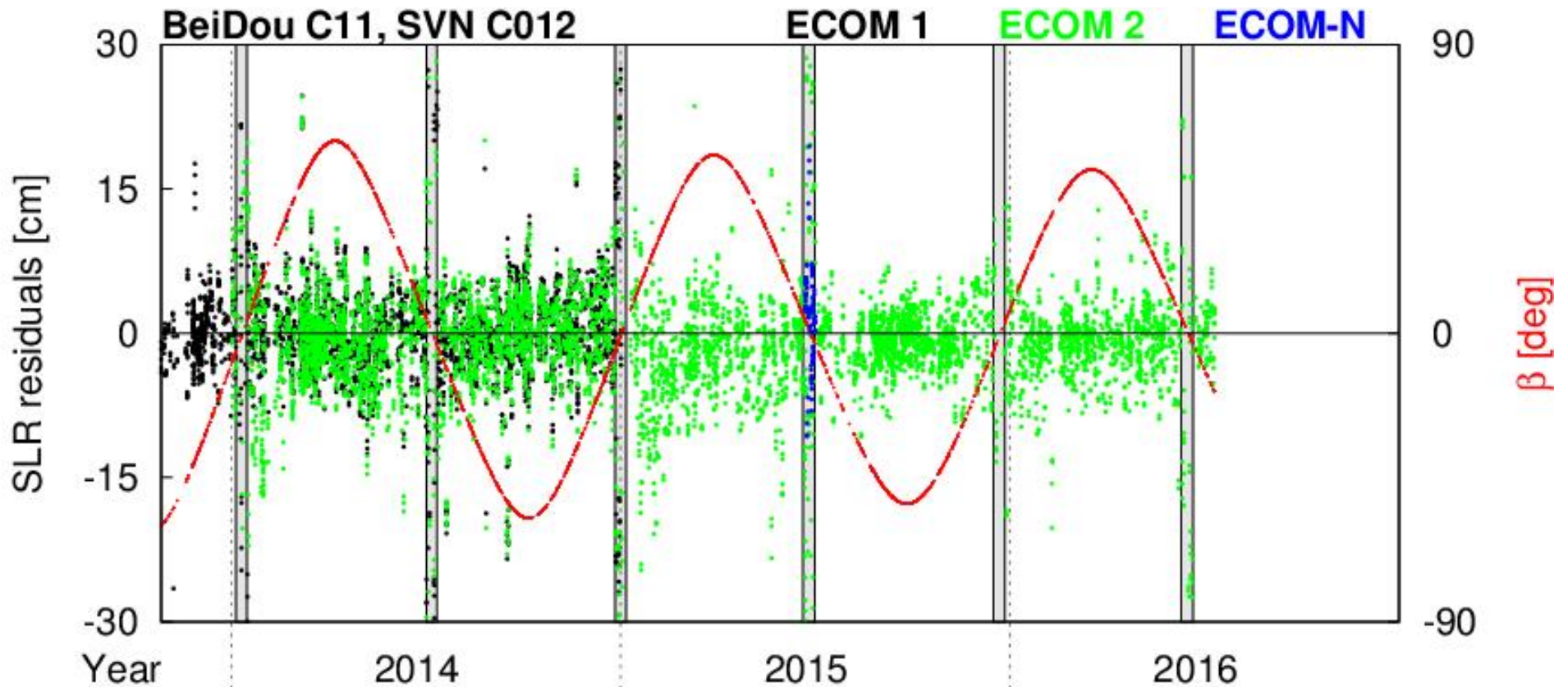
# BeiDou orbit validation



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- ⇒ Yaw-steering: no significant impact of ECOM version
- ⇒ Orbit normal attitude mode ( $|\beta| < 4^\circ$ ; grey boxes; not correctly considered): large residuals

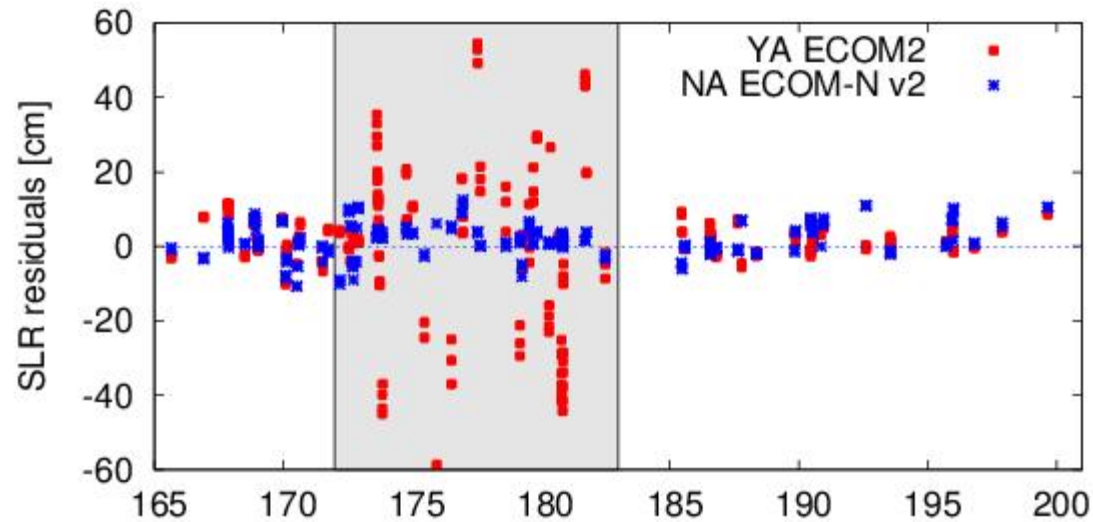
# BeiDou orbit validation



⇒ Test of new ECOM versions better suited for orbit normal attitude mode

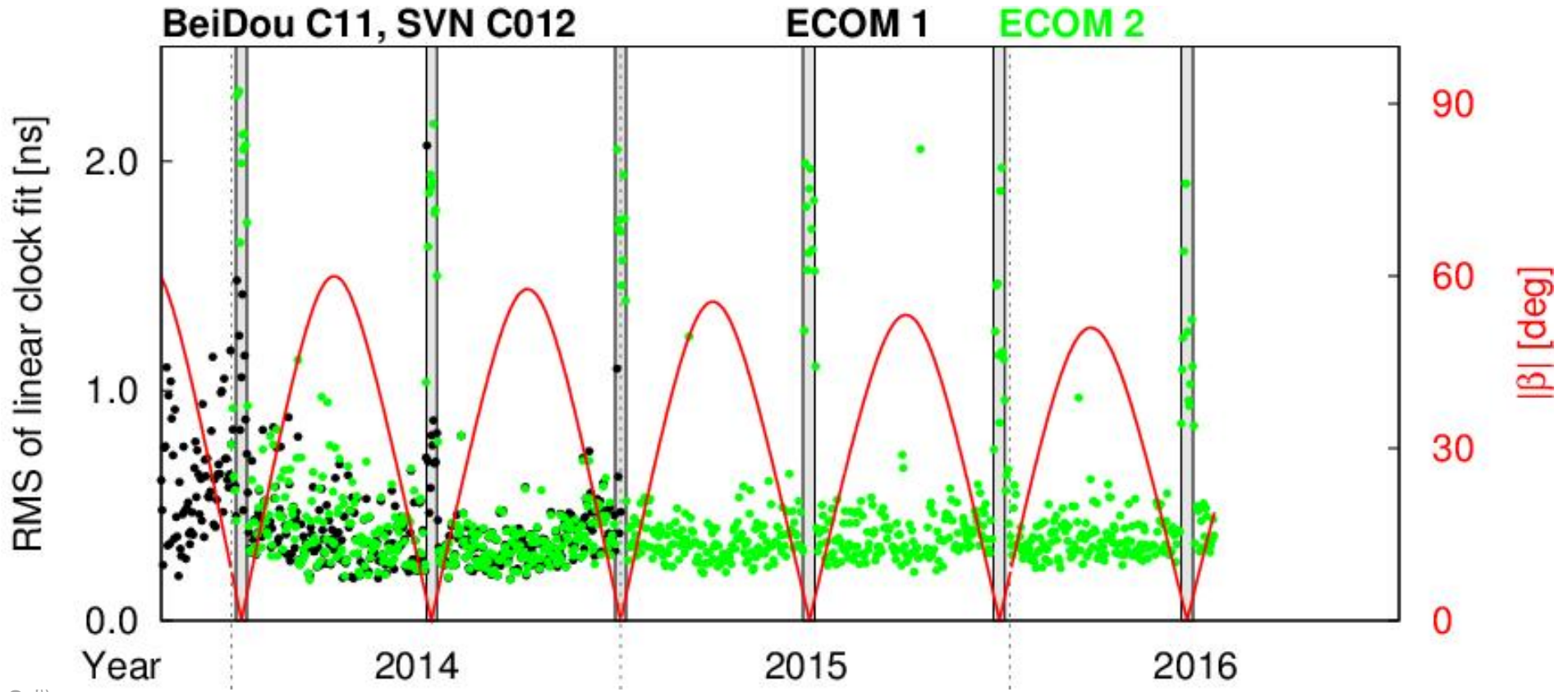
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# BeiDou orbit validation



⇒ Test of new ECOM versions better suited for orbit normal attitude mode

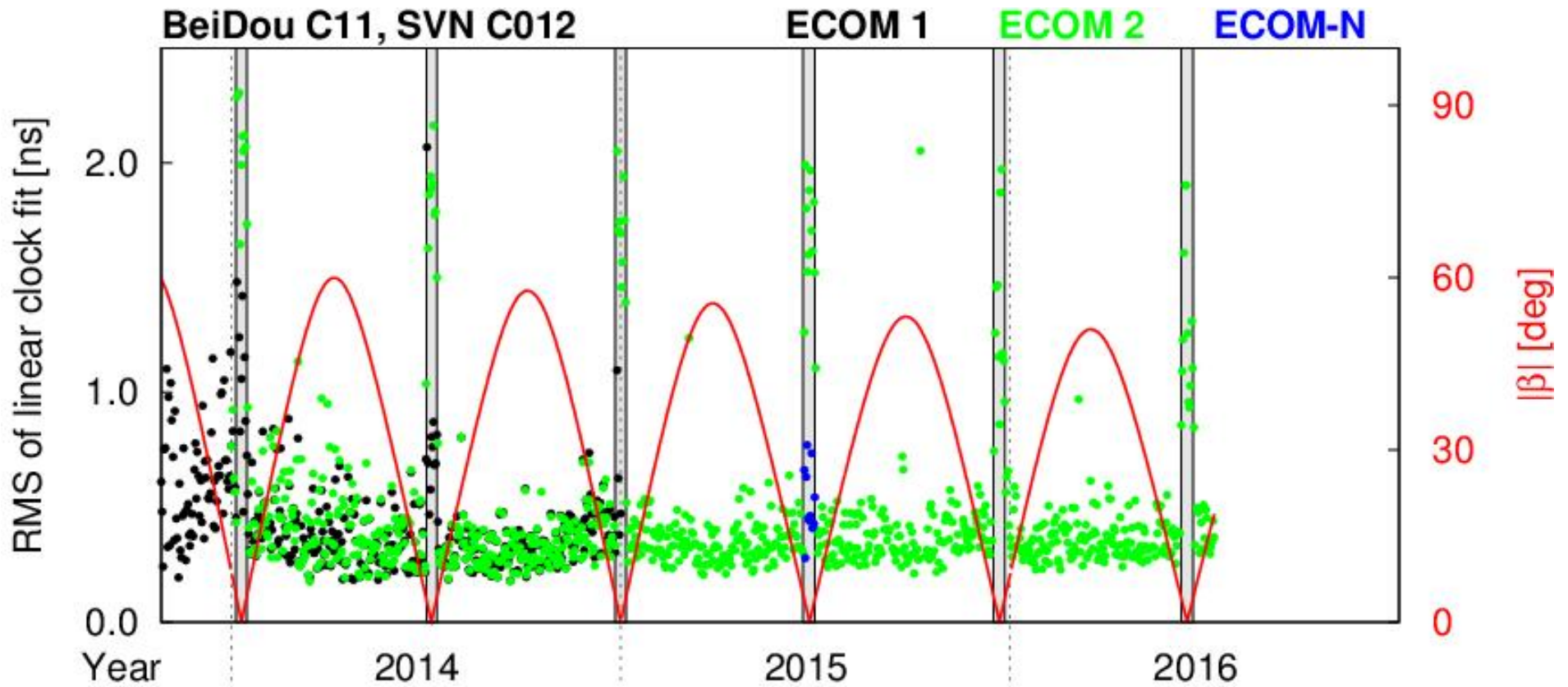
# BeiDou clock validation



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- ⇒ Yaw-steering: no significant difference between ECOM versions
- ⇒ Orbit normal attitude mode ( $|\beta| < 4^\circ$ , marked grey; wrong attitude considered): ECOM2 may even degrade solution

# BeiDou clock validation

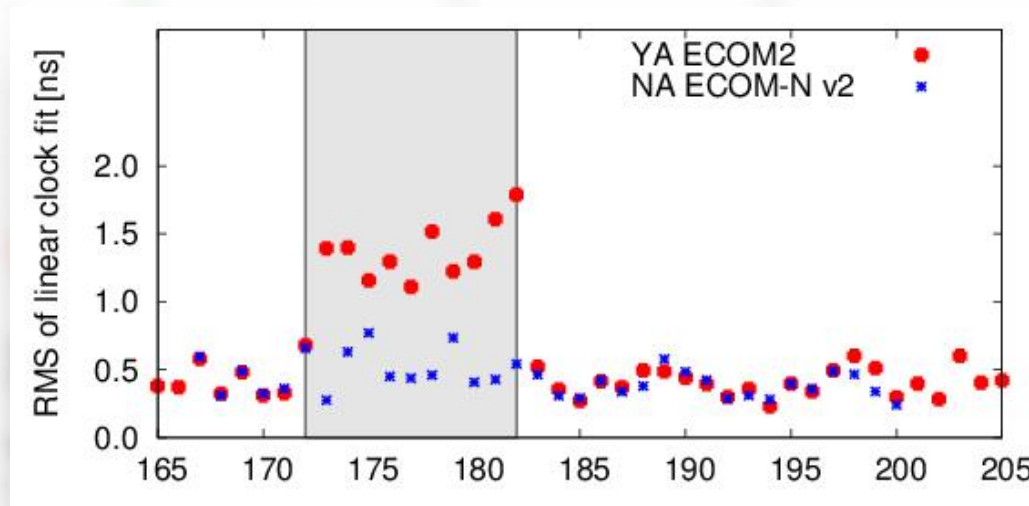


⇒ Experiments with ECOM versions better suited for orbit normal attitude mode

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# BeiDou clock validation

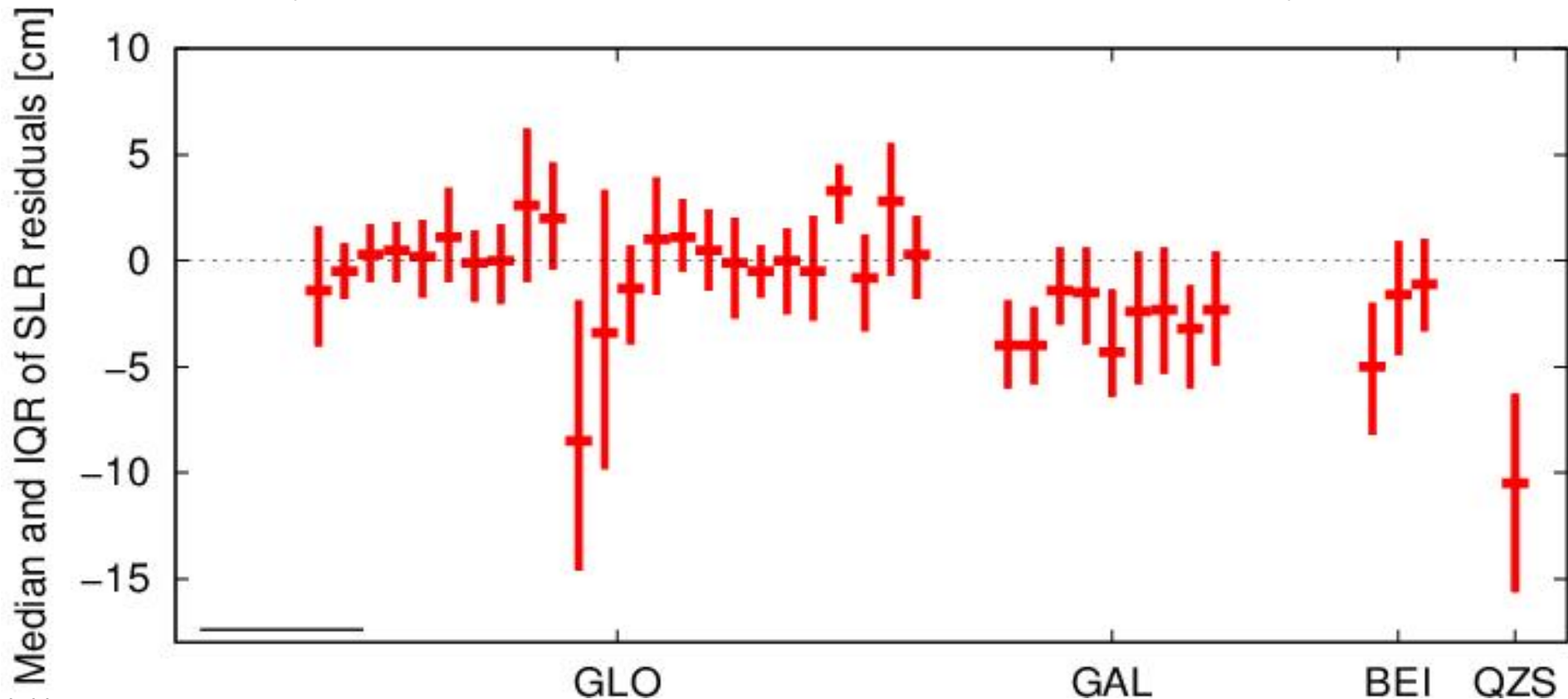


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⇒ Experiments with ECOM versions better suited for orbit normal attitude mode

# Orbit validation 2015: SLR

(Median and IQR; satellites in eclipse or normal mode are not considered)

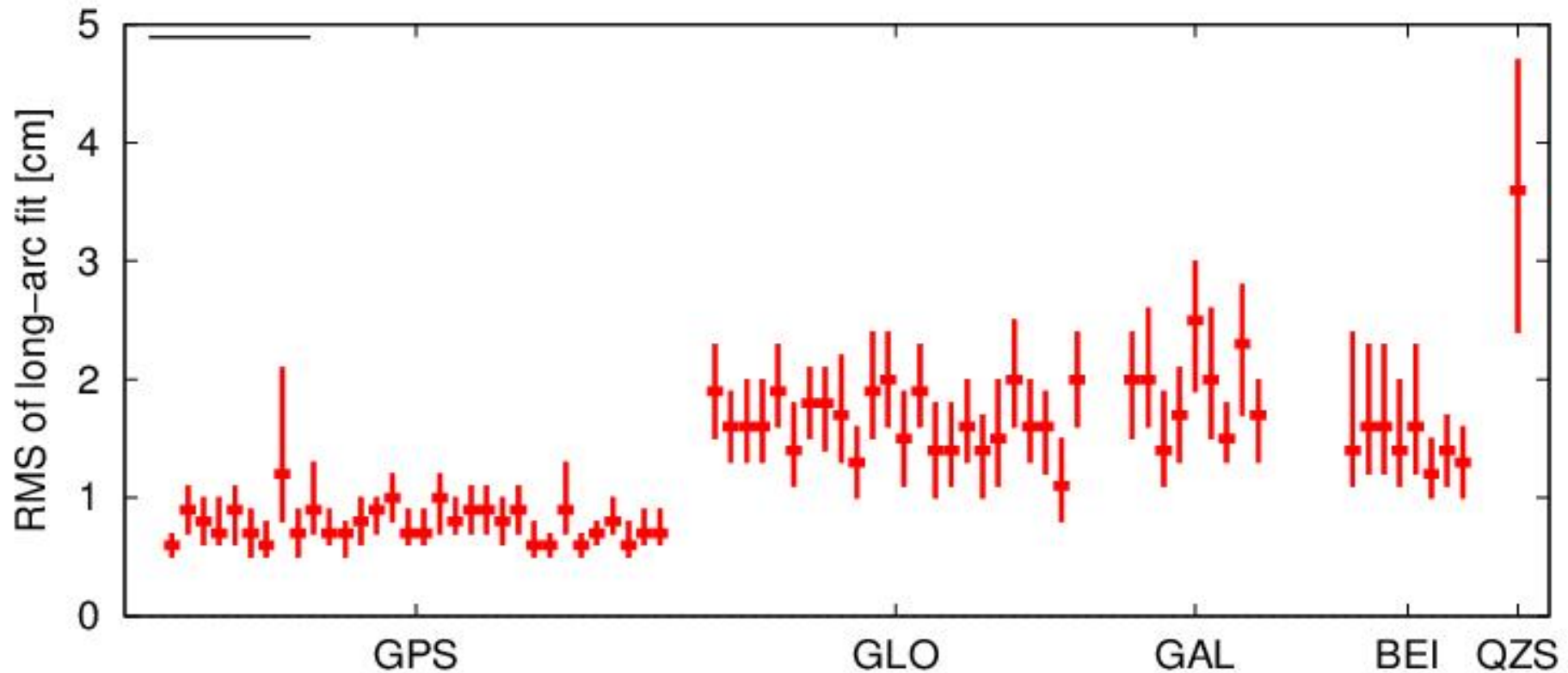


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- ⇒ GLONASS: inhomogeneous (unknown issues with some SCs)
- ⇒ New systems: significant offsets due to modelling deficiencies (e.g., ANTEX, albedo, antenna thrust)

# Orbit validation 2015: 3-day long-arc fit

(Median and IQR; satellites in eclipse or normal mode are not considered)

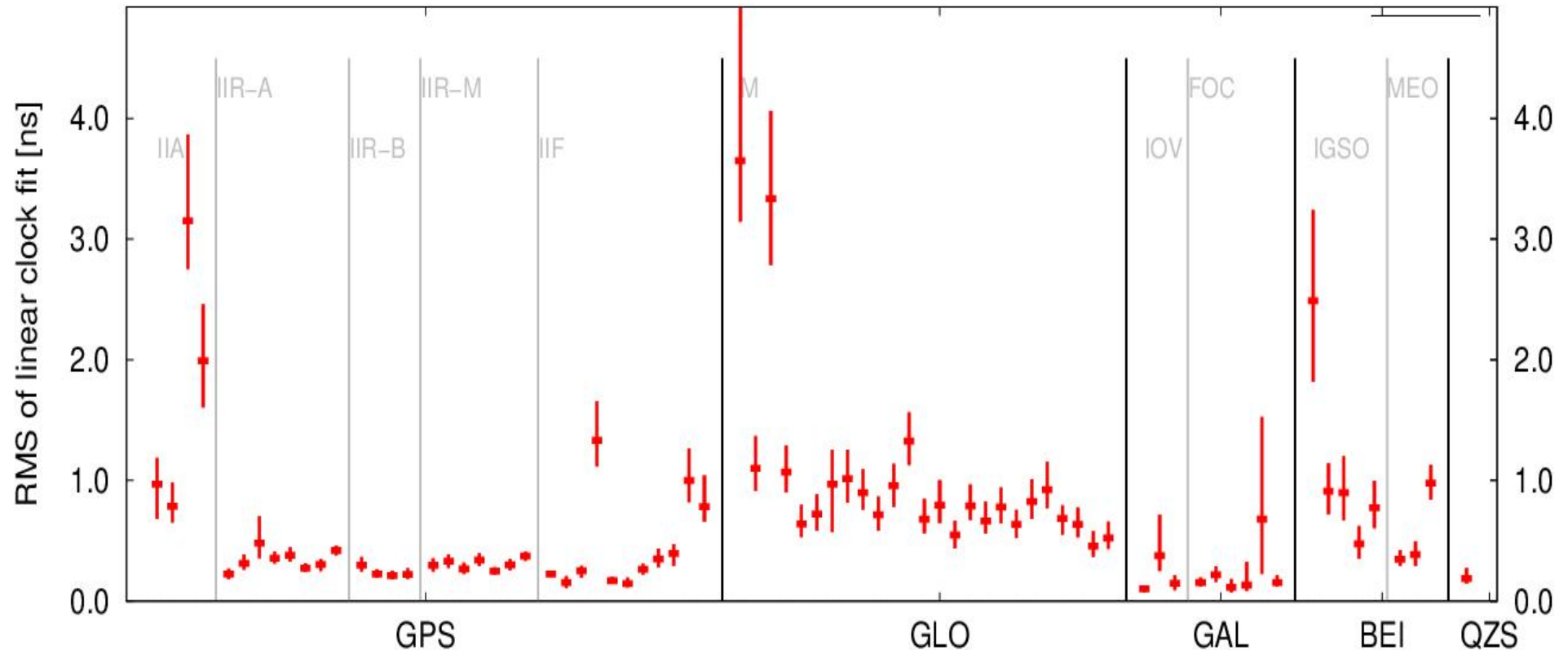


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- ⇒ Galileo and BeiDou: almost comparable to GLONASS
- ⇒ QZSS performs worst
- ⇒ BeiDou MEOs better than IGSOs

# Clock validation 2015: median RMS of daily linear fit

(Median and IQR; satellites in eclipse or normal mode are not considered)



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- ⇒ Galileo PHM, QZS-1, most GPS IIR and IIF: excellent performance
- ⇒ GPS IIA, some IIF, GLONASS, Galileo RAFS: less good (RMS: 0.5 ns or bigger)
- ⇒ BeiDou: mixed performance

# COM to-do list

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- Implementation of Galileo, QZSS, BeiDou (except GEOs)
- Use of RINEX3 files from IGS and EPN - now also with long file names; selection of observation types
- Improved SRP model for yaw-steering attitude (ECOM2, Arnold et al., 2015)
- Normal attitude and related SRP models for QZSS and BeiDou
- Attitude laws for GPS, GLONASS, (Galileo?) eclipses
- ANTEX (PCO+PCV) for Galileo, QZSS, BeiDou
- Proper handling of observation biases; new BIAS-SINEX version
- Ambiguity resolution for Galileo, BeiDou, QZSS; improvement for GLONASS
- Albedo radiation modelling for Galileo, QZSS, BeiDou
- Antenna thrust for (GLONASS), Galileo, QZSS, BeiDou

# Summary

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- Still a some way to go until the new GNSS can contribute to CODE's IGS solutions to the same extent as GPS
- Our current focus: orbit normal attitude (challenges are, e.g., SRP modelling, detection and consideration of mode-transitions, ...), correct handling of observation biases, PCO+PCV estimation for new systems
- Reprocessing of data from 2015 or 2016 planned after implementation of further improvements

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Thank you  
for  
your interest!